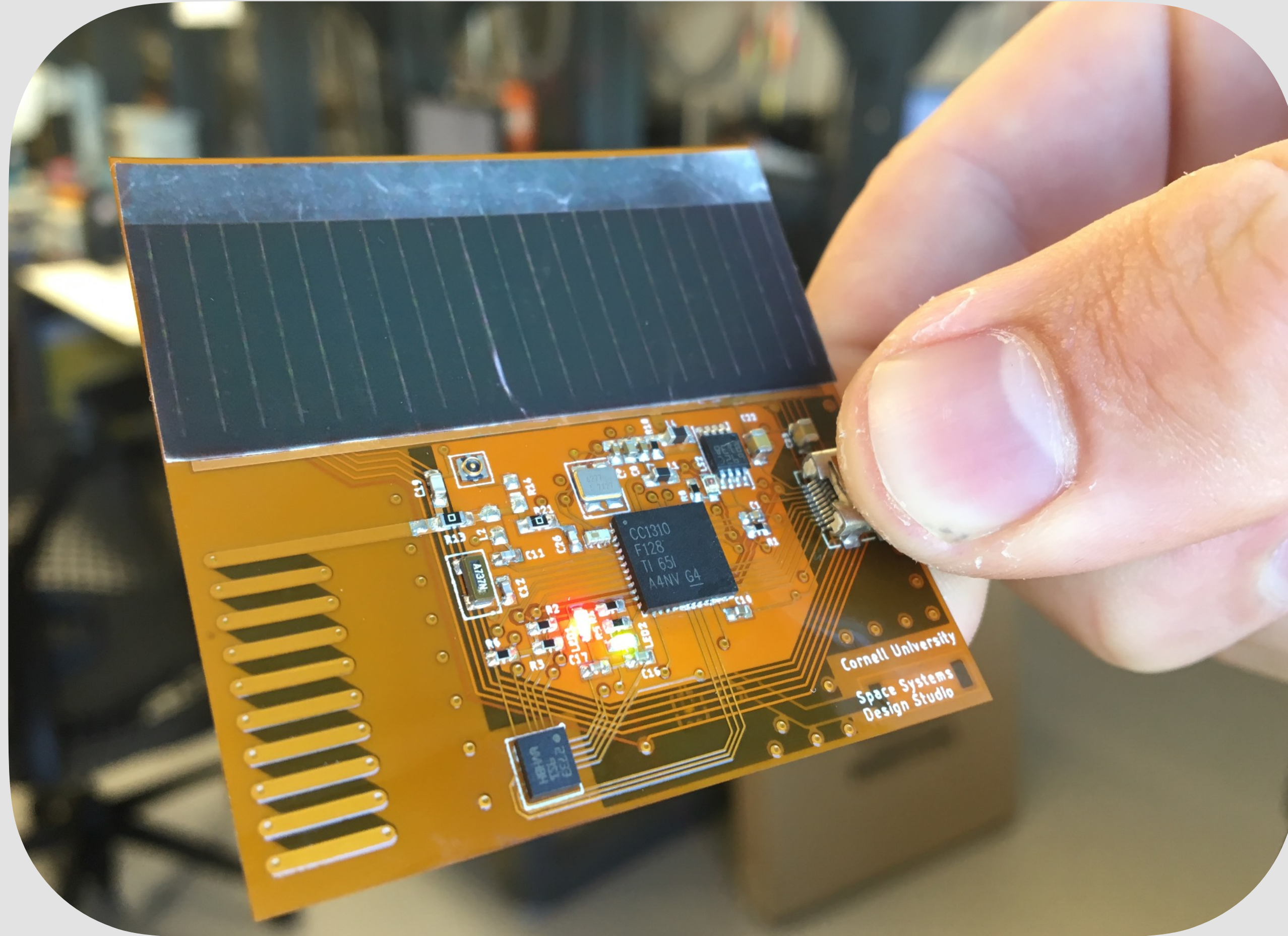
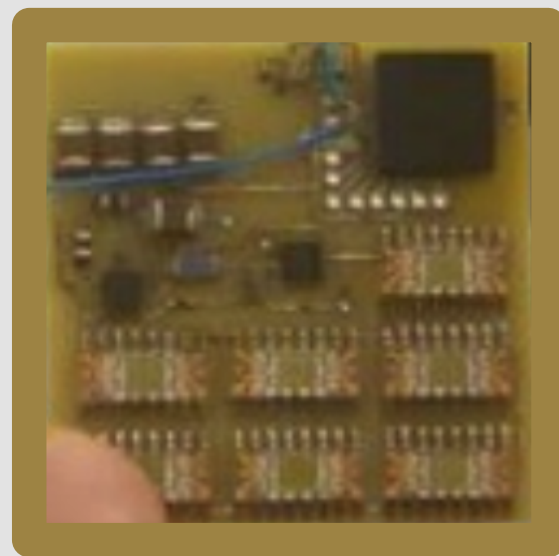


History, State of the Art, and Future of Gram-Scale Spacecraft

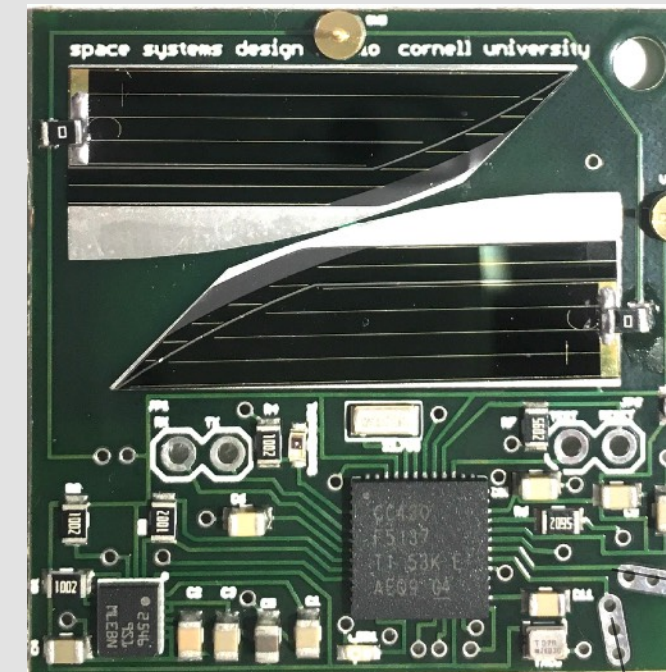


V. Hunter Adams
September 29, 2018
Bremen, Germany

1. (Very) brief history of the hardware
2. Description of the state of the art
3. Short-term mission and technology possibilities
4. Intermediate-term mission and technology possibilities
5. Long-term speculation



International Space Station Demo



KickSat 2
Alpha

2010

2016

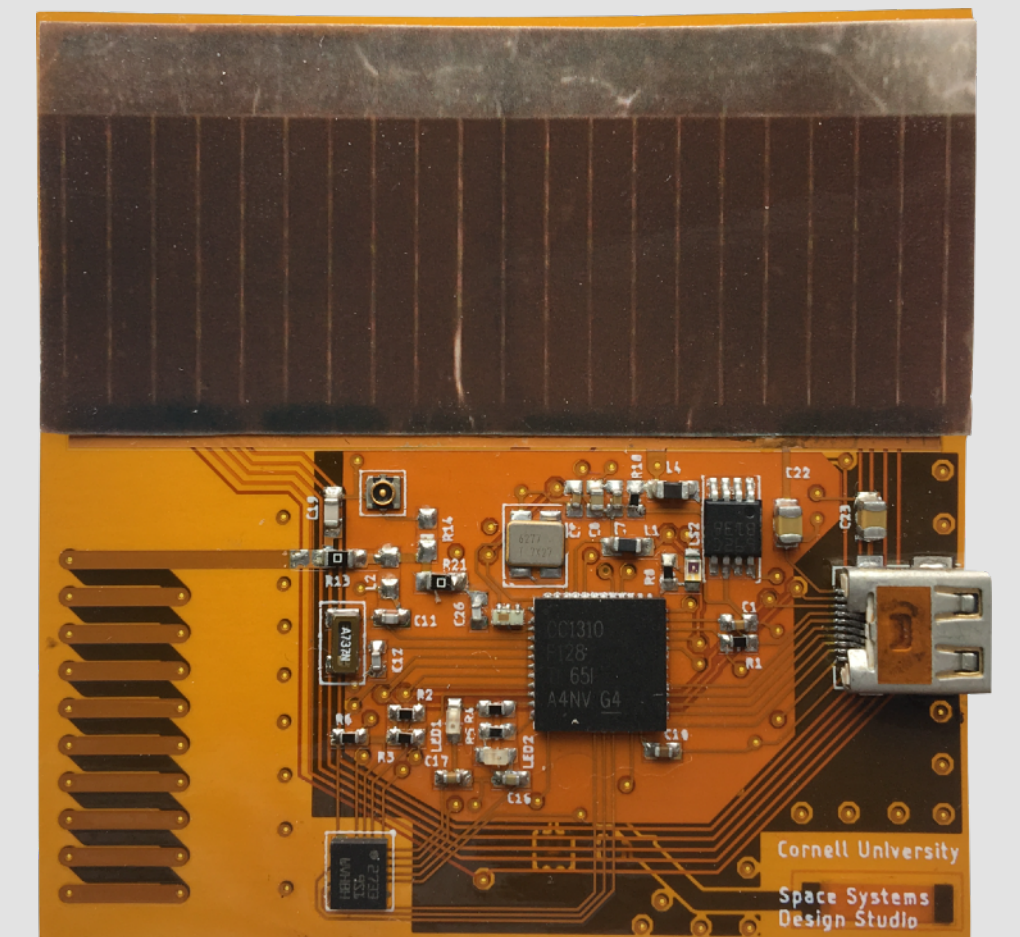
2013

2018

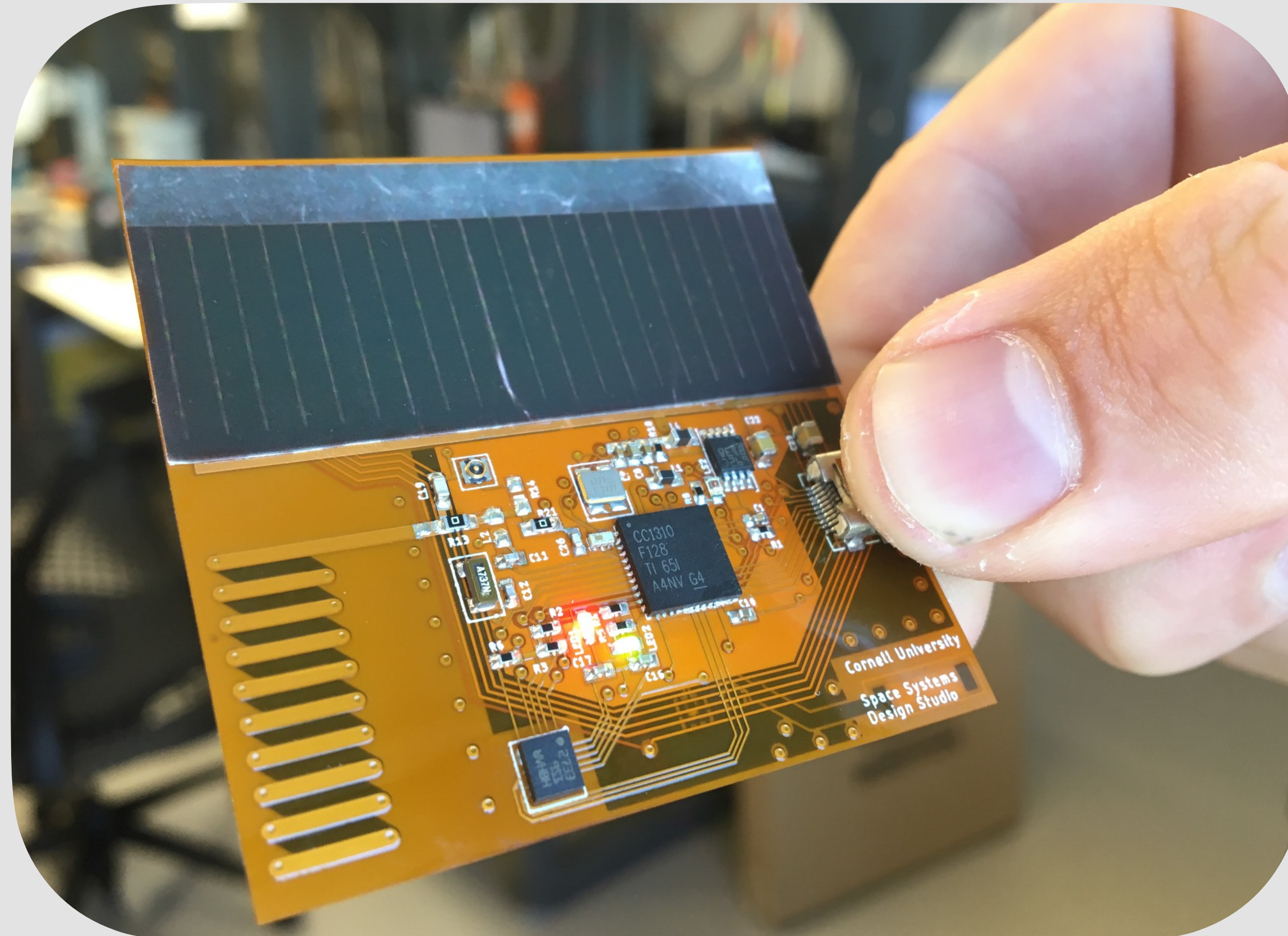
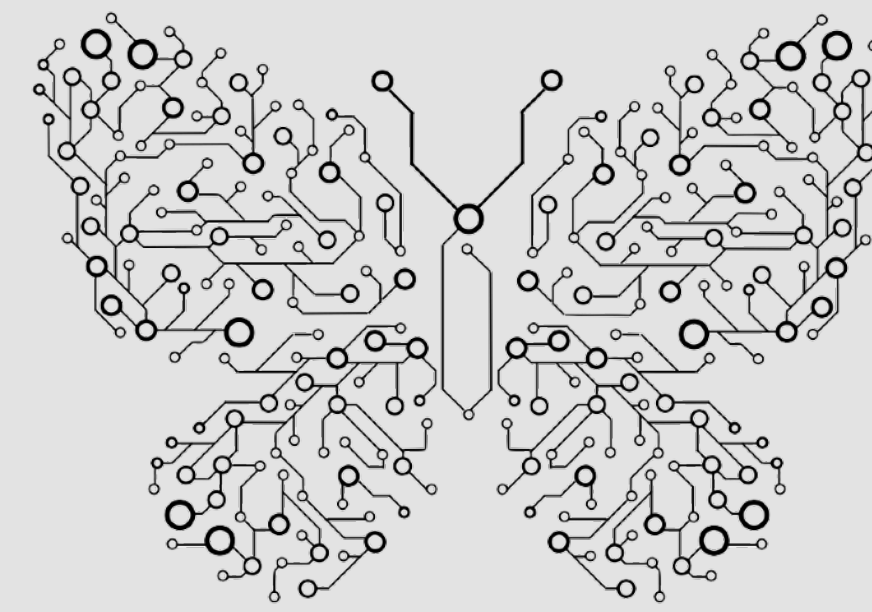
KickSat 1,2
Venta 1

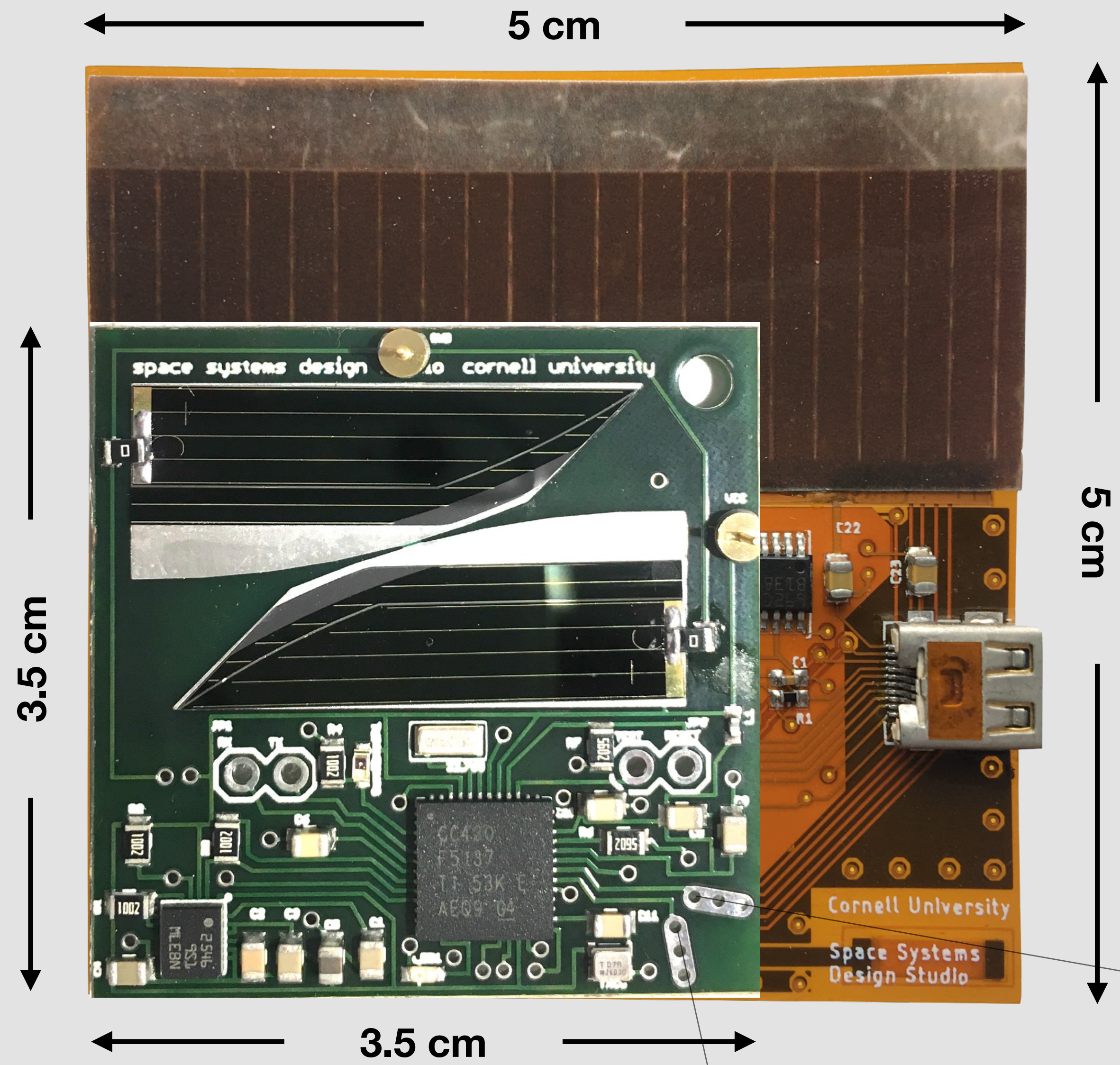


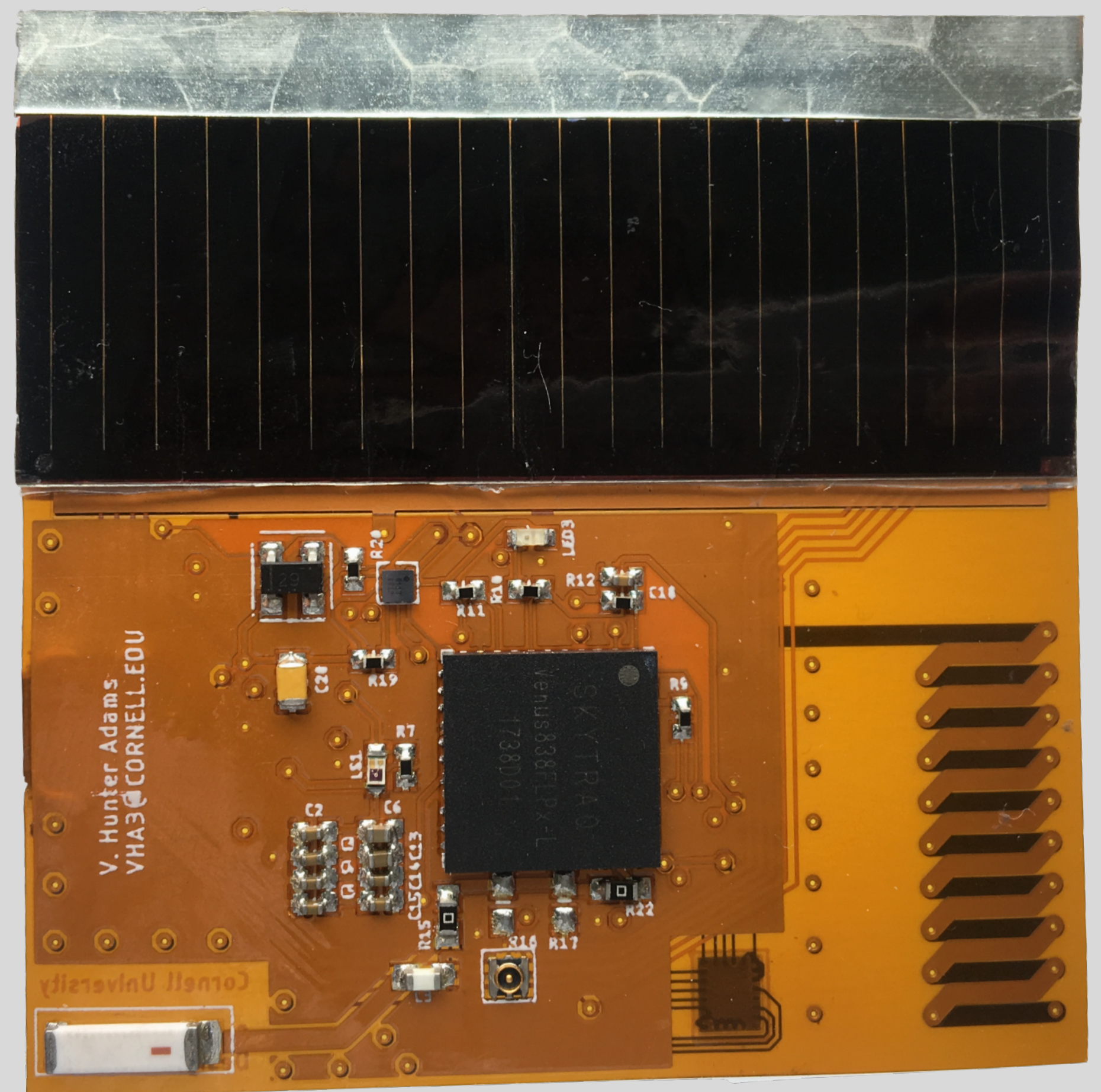
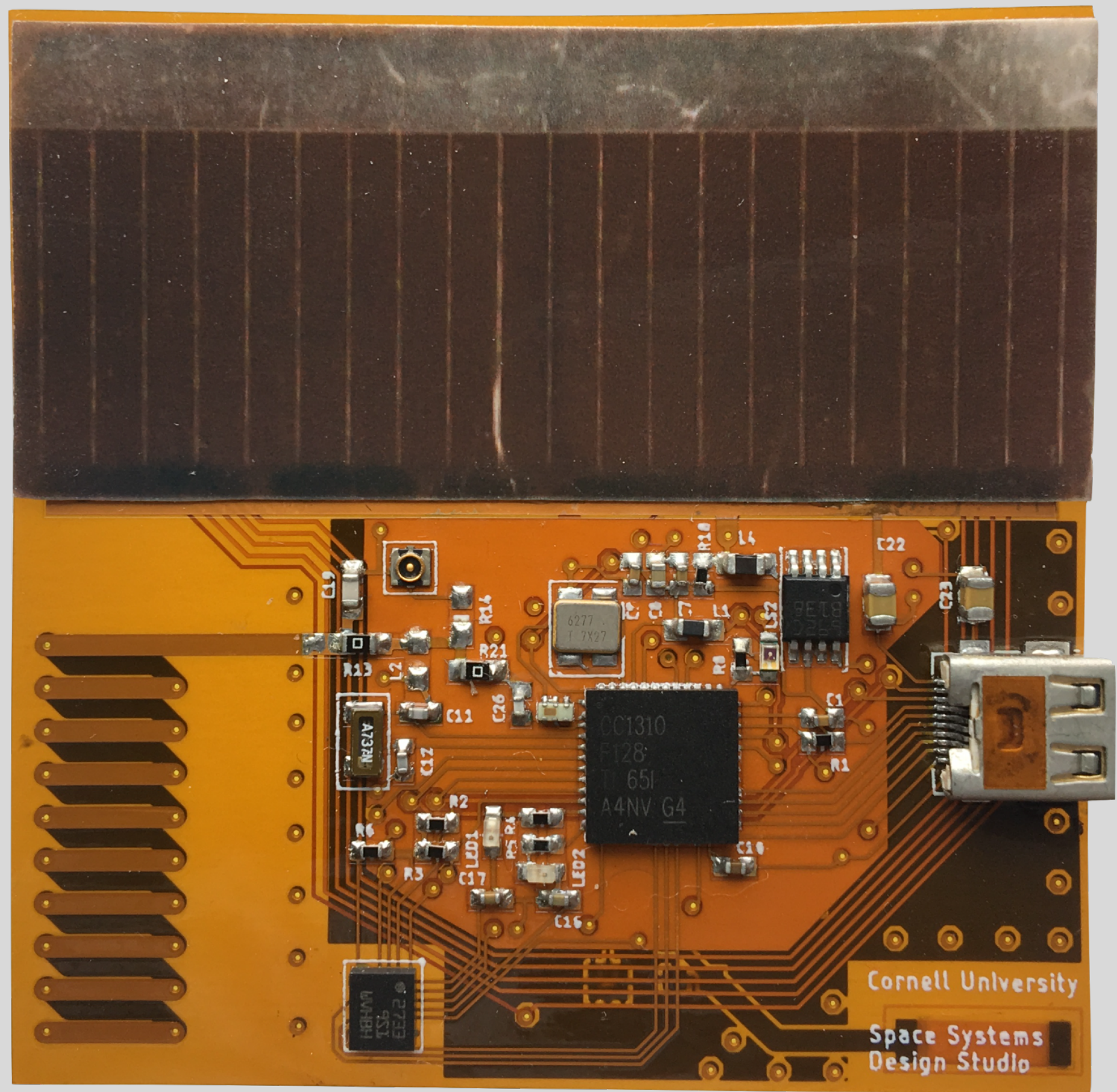
Proposals out

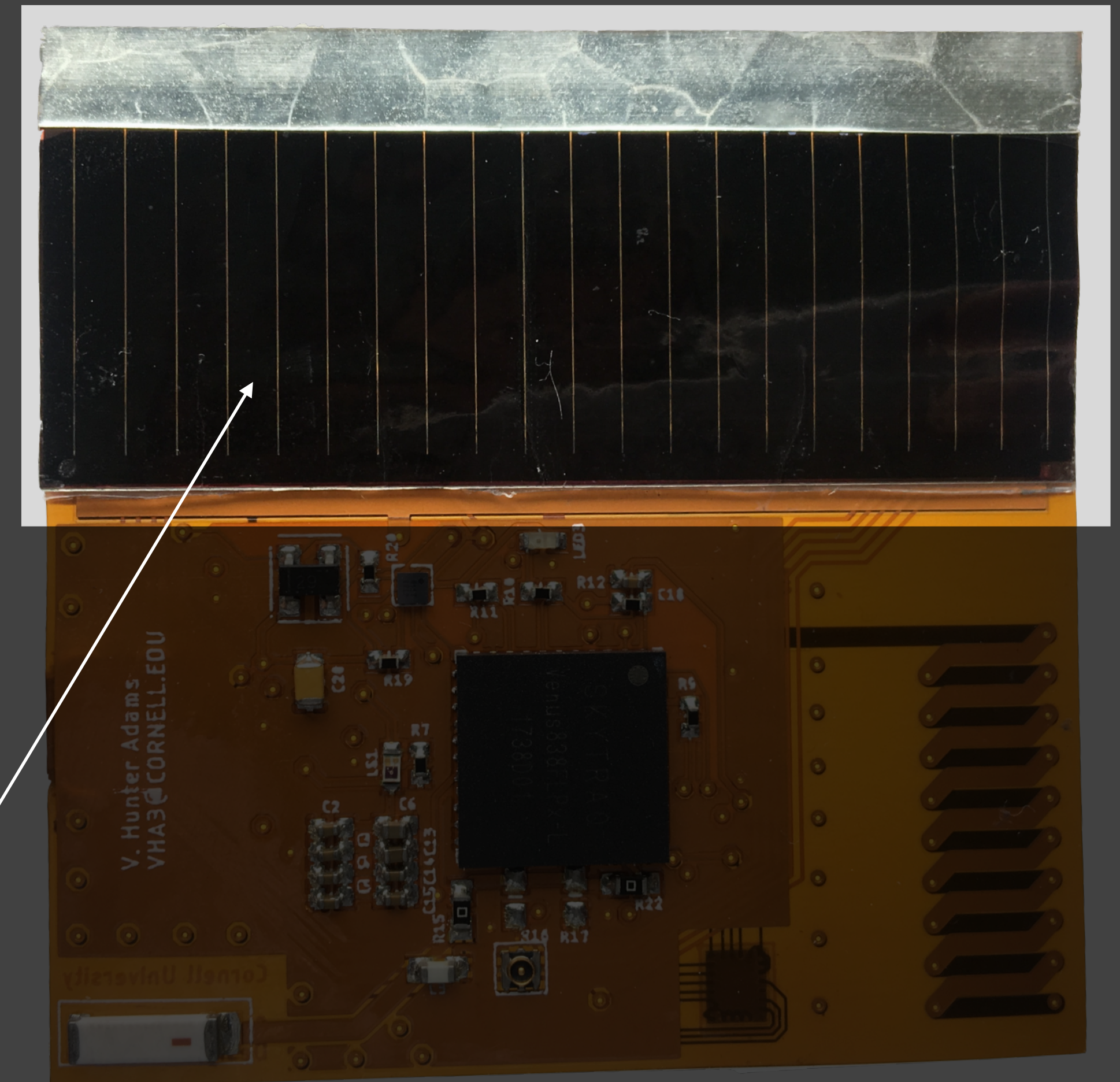
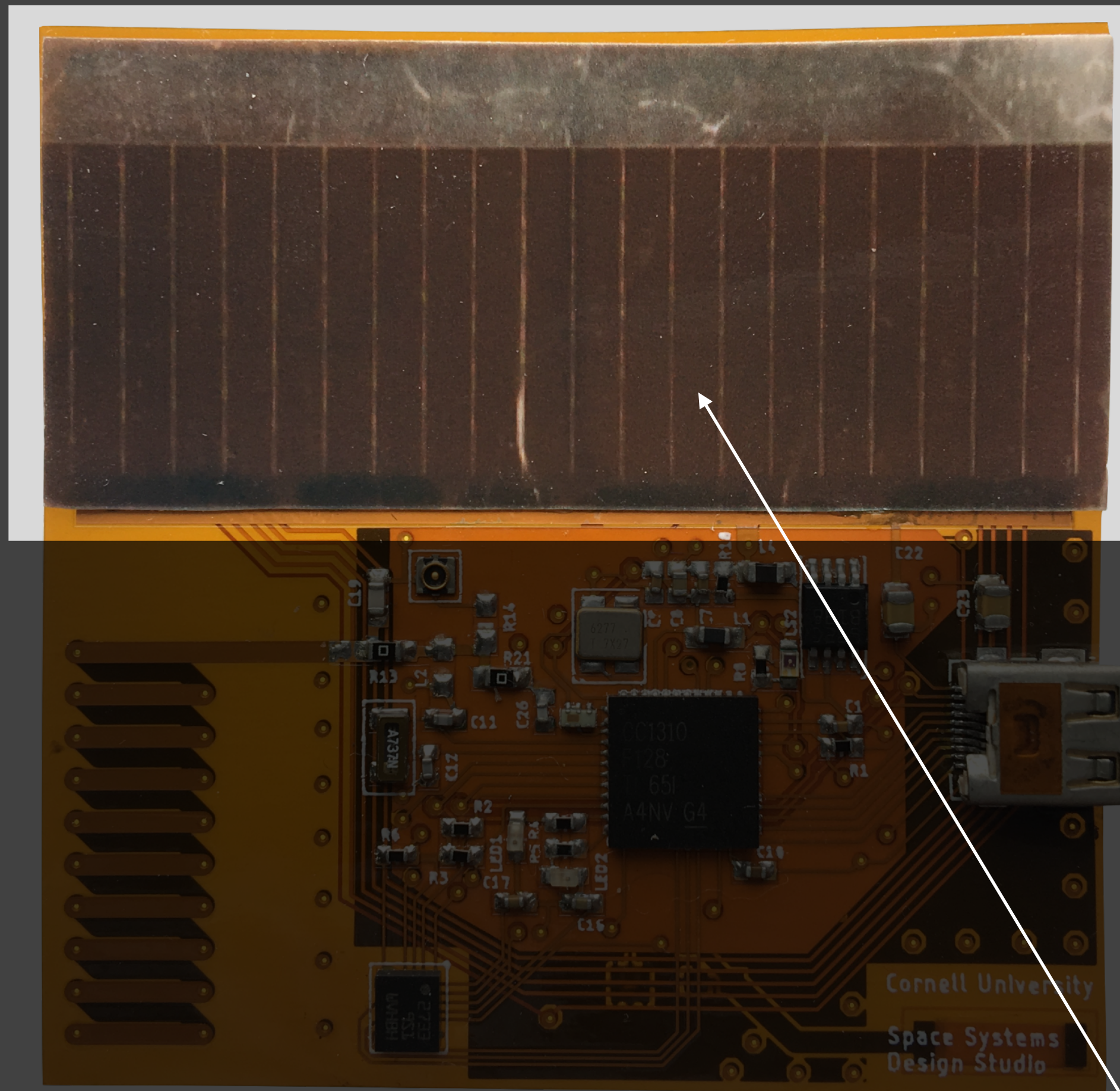


Monarch

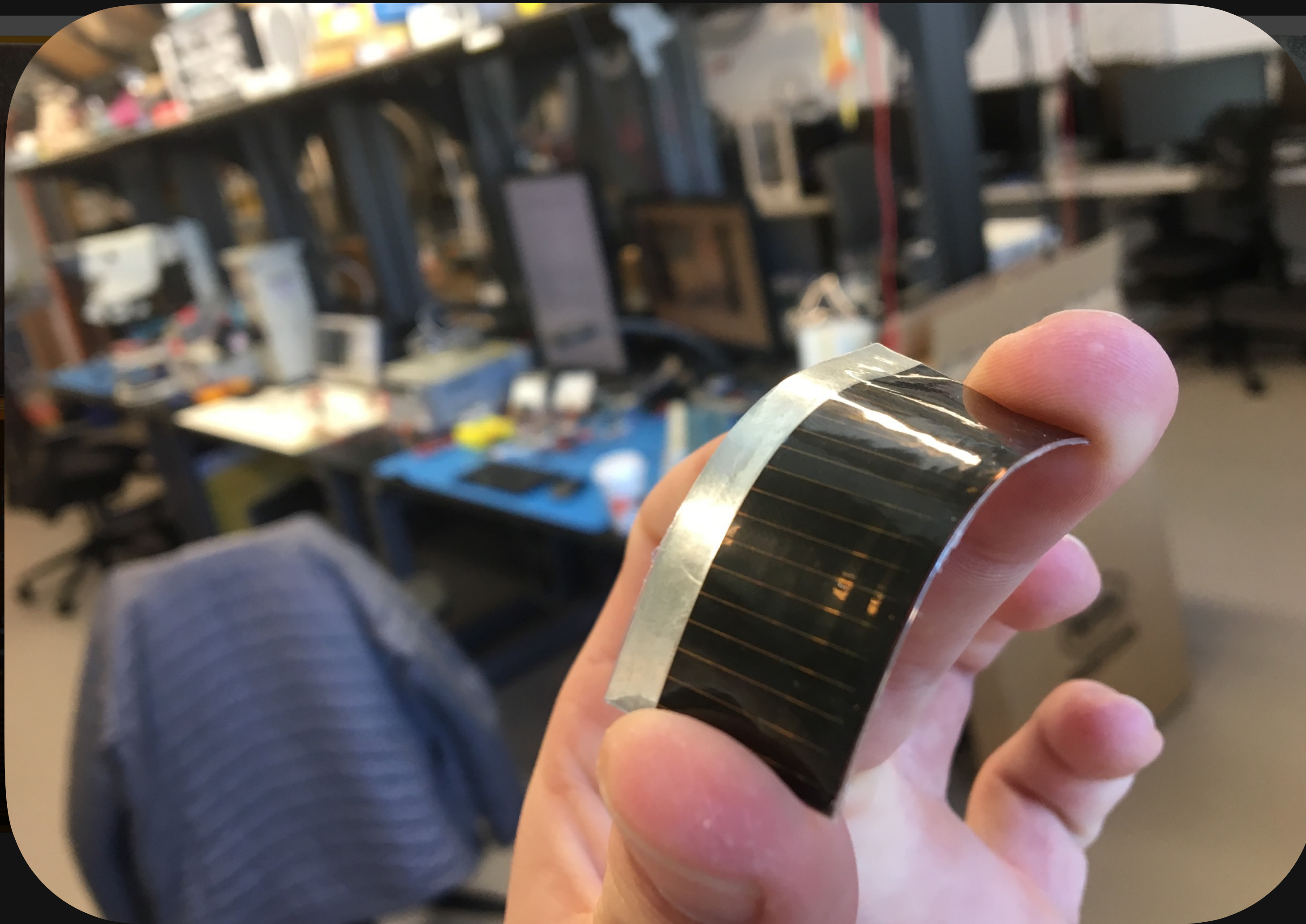




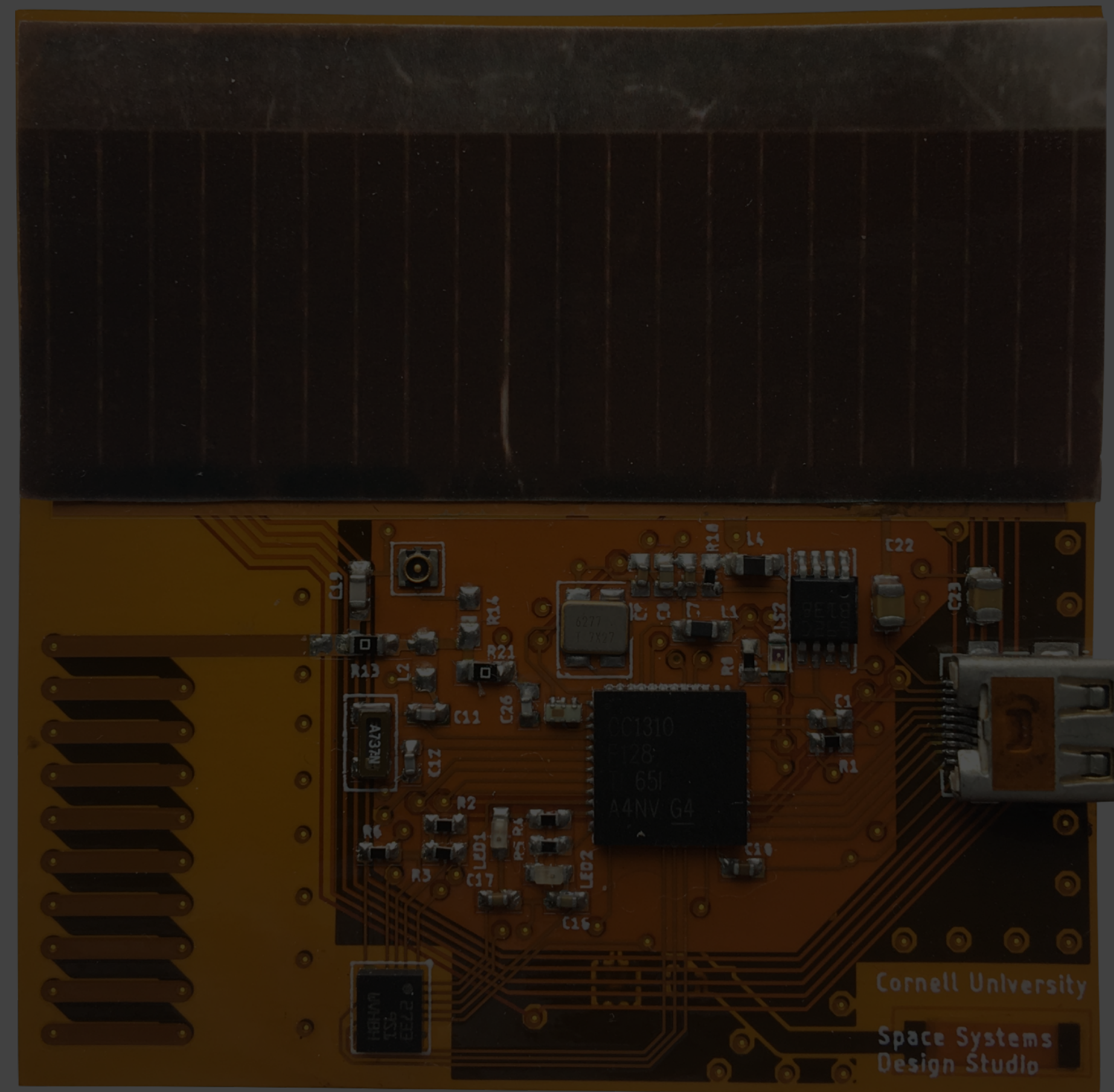




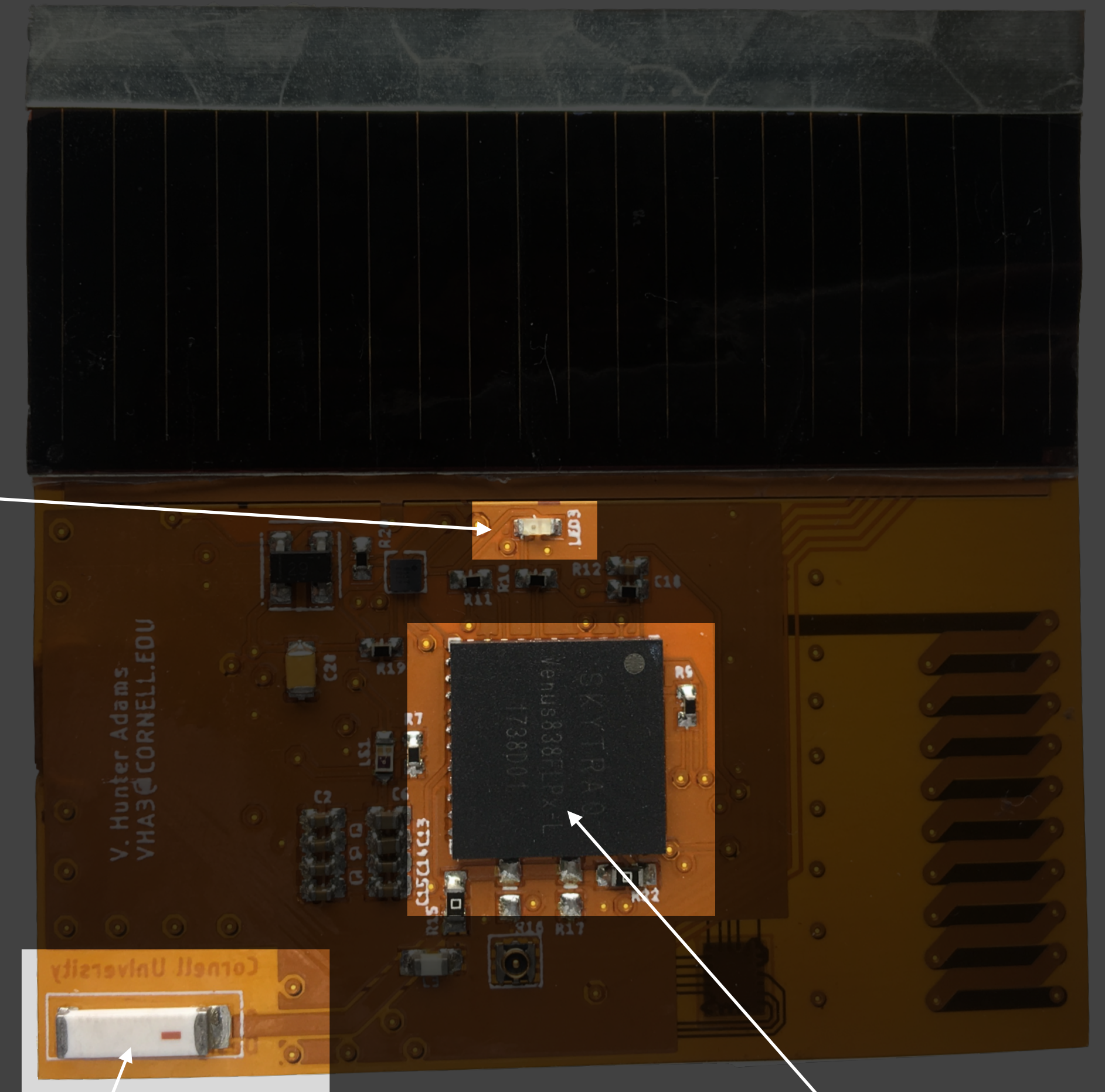
Alta Devices solar cells - 300 mW each



0.1 grams, and flexible

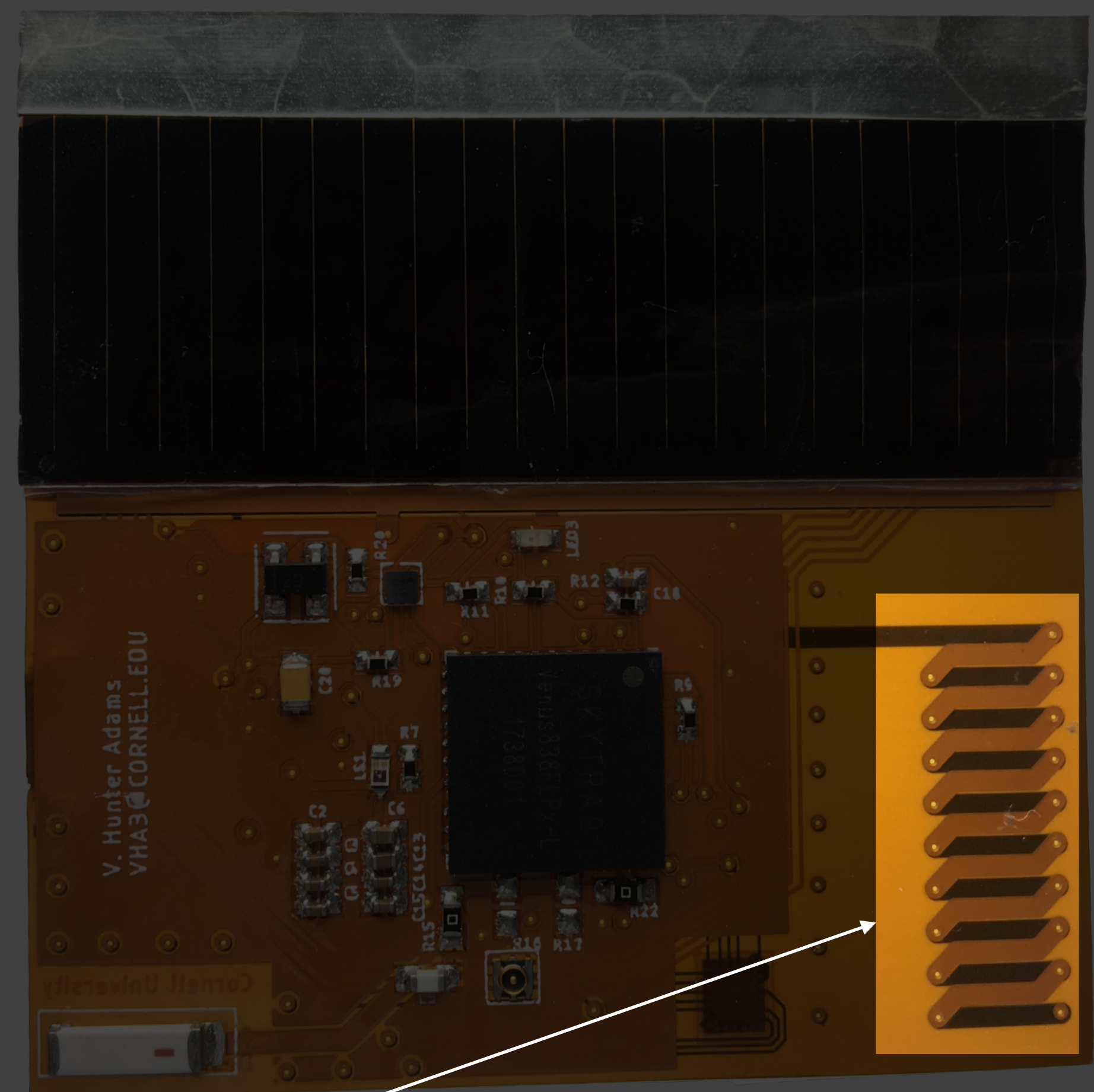
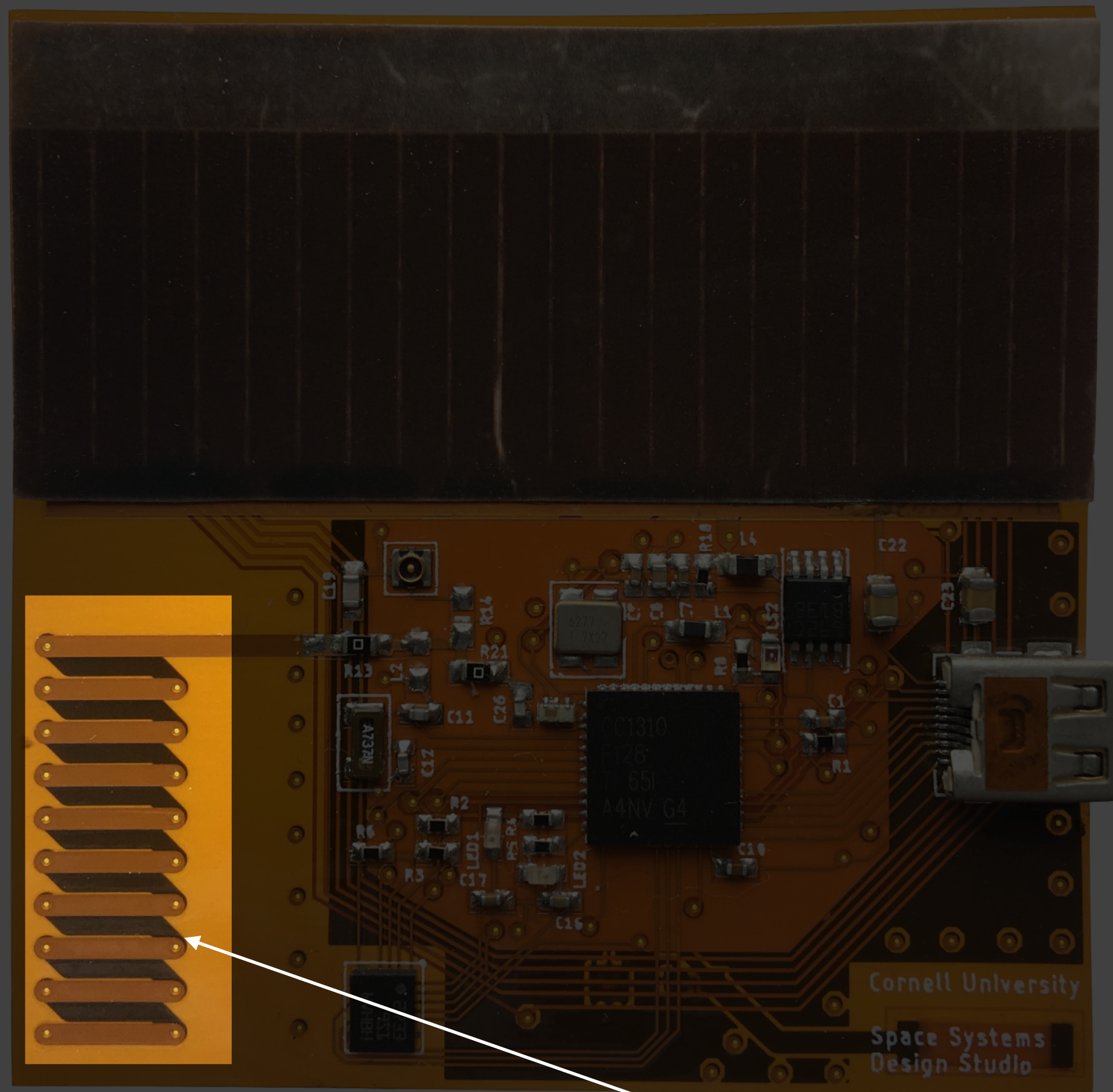


GPS fix indicator

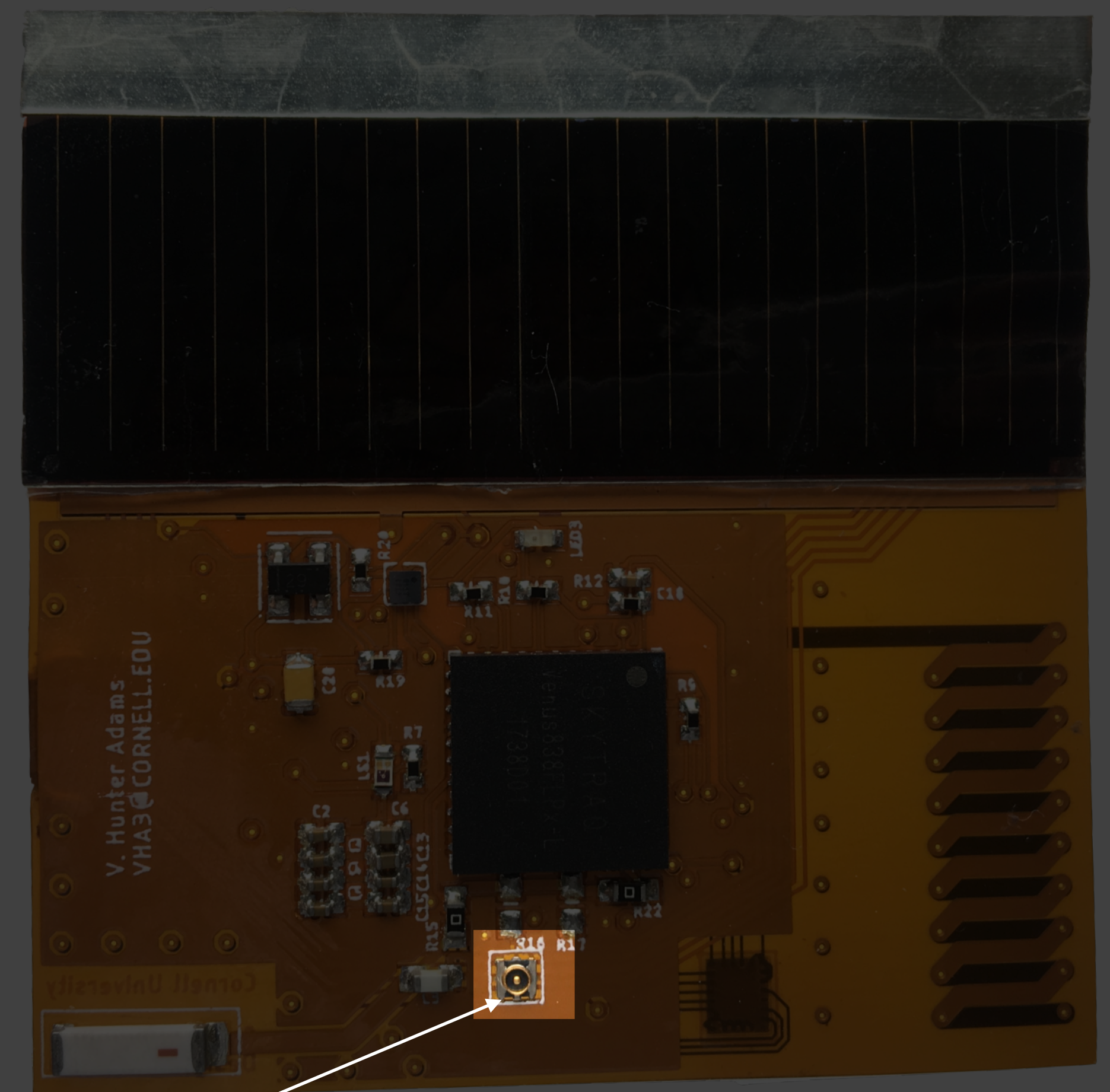
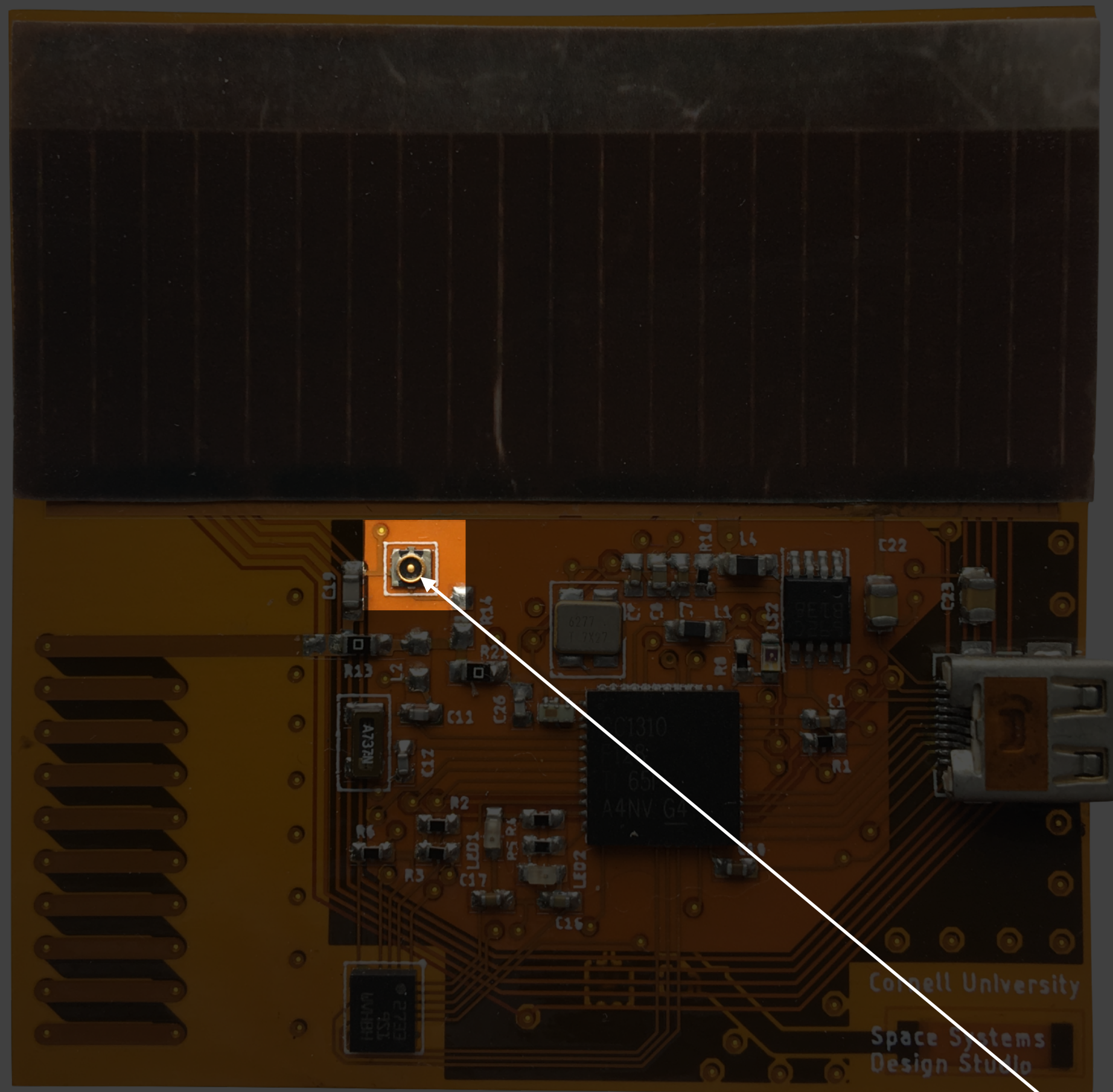


GPS antenna

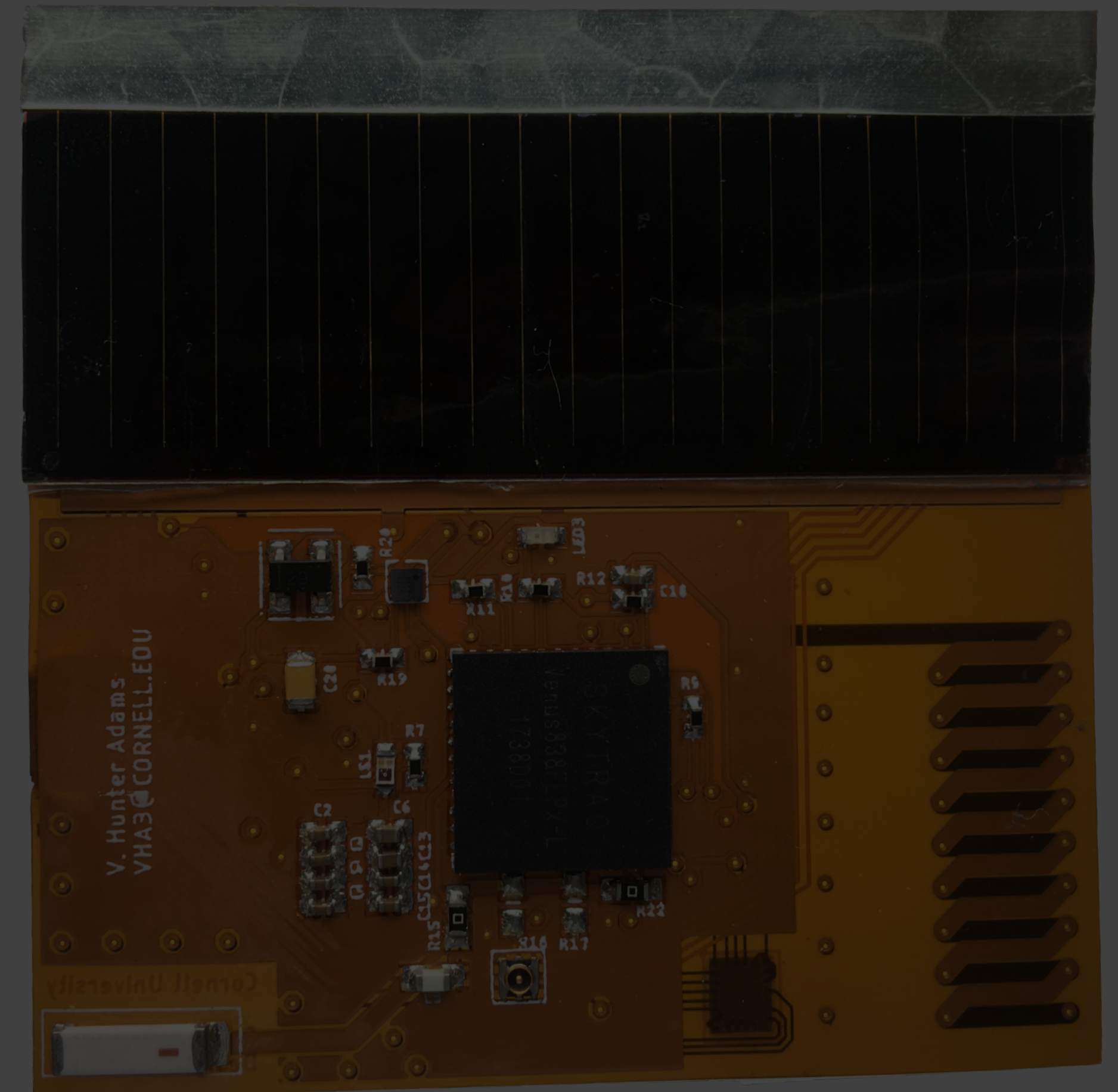
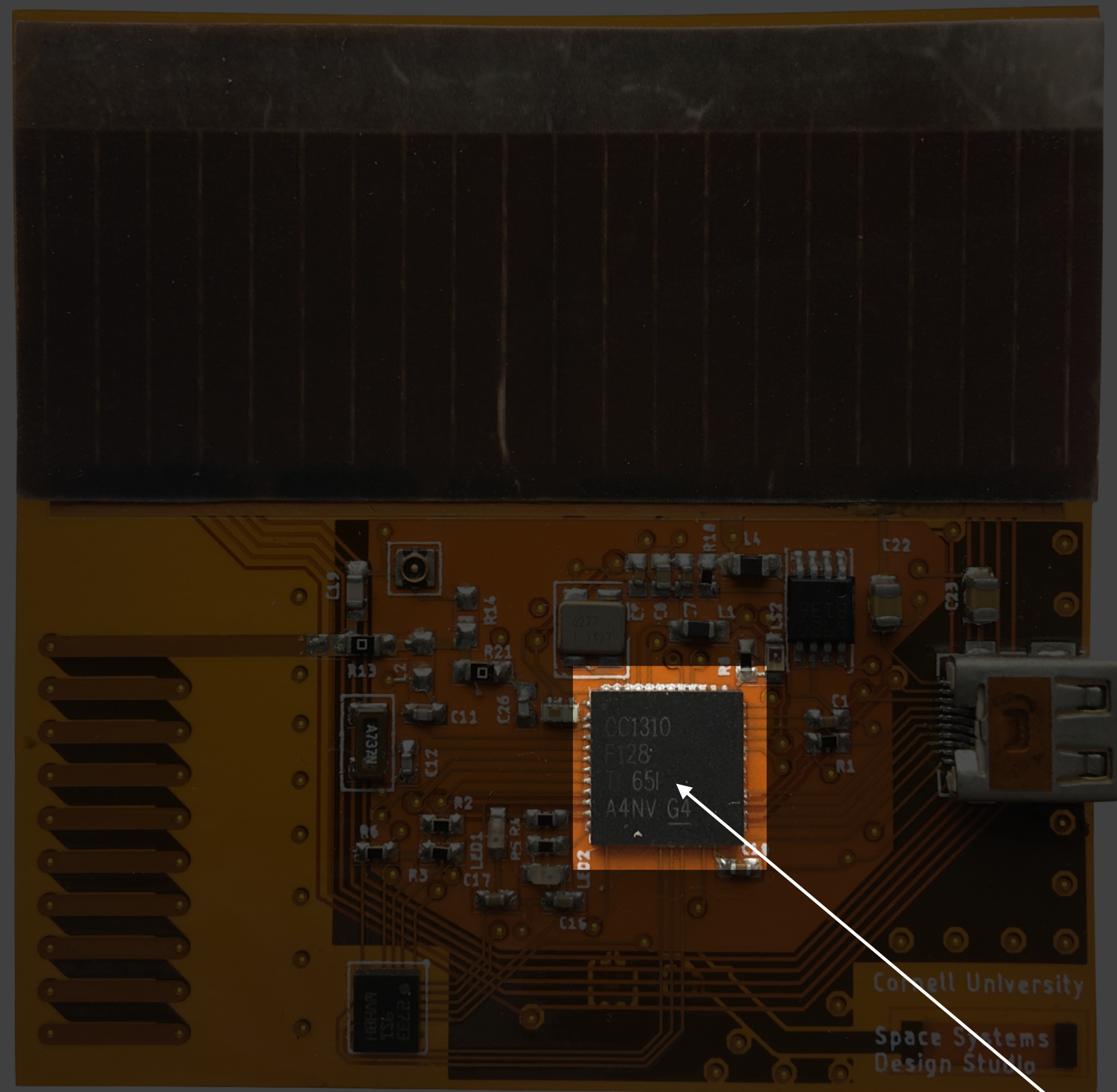
GPS module



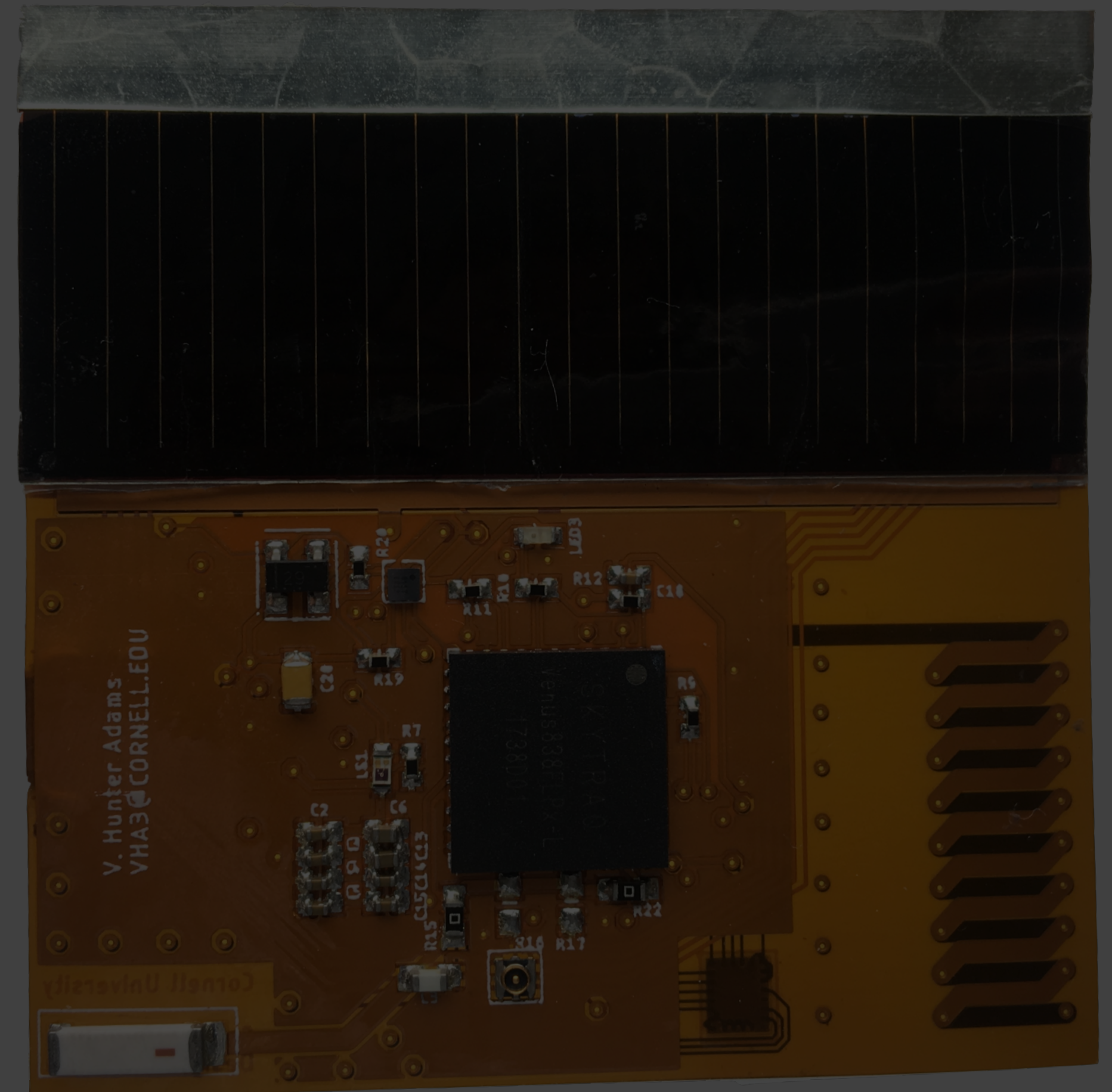
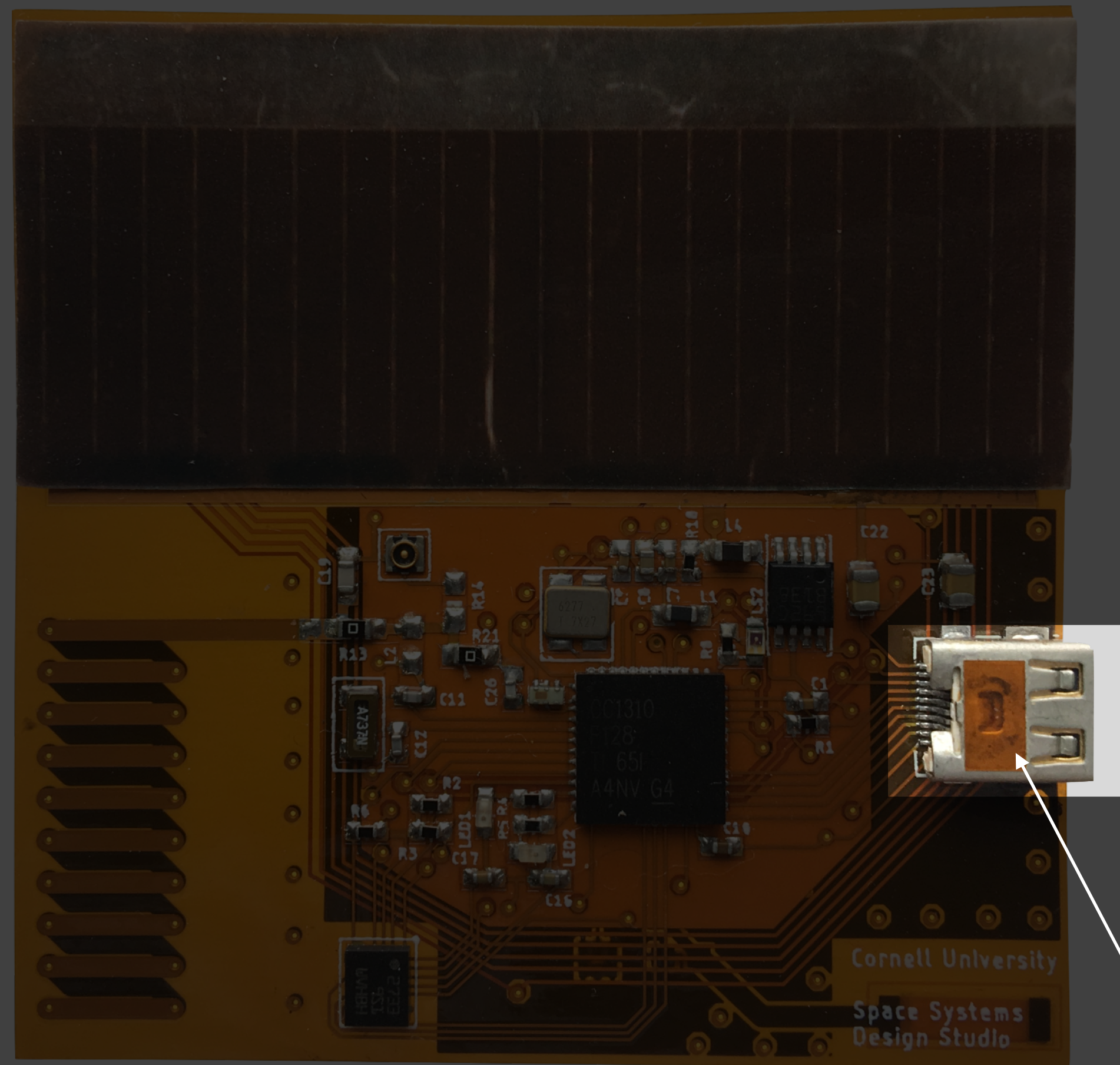
embedded PCB antenna



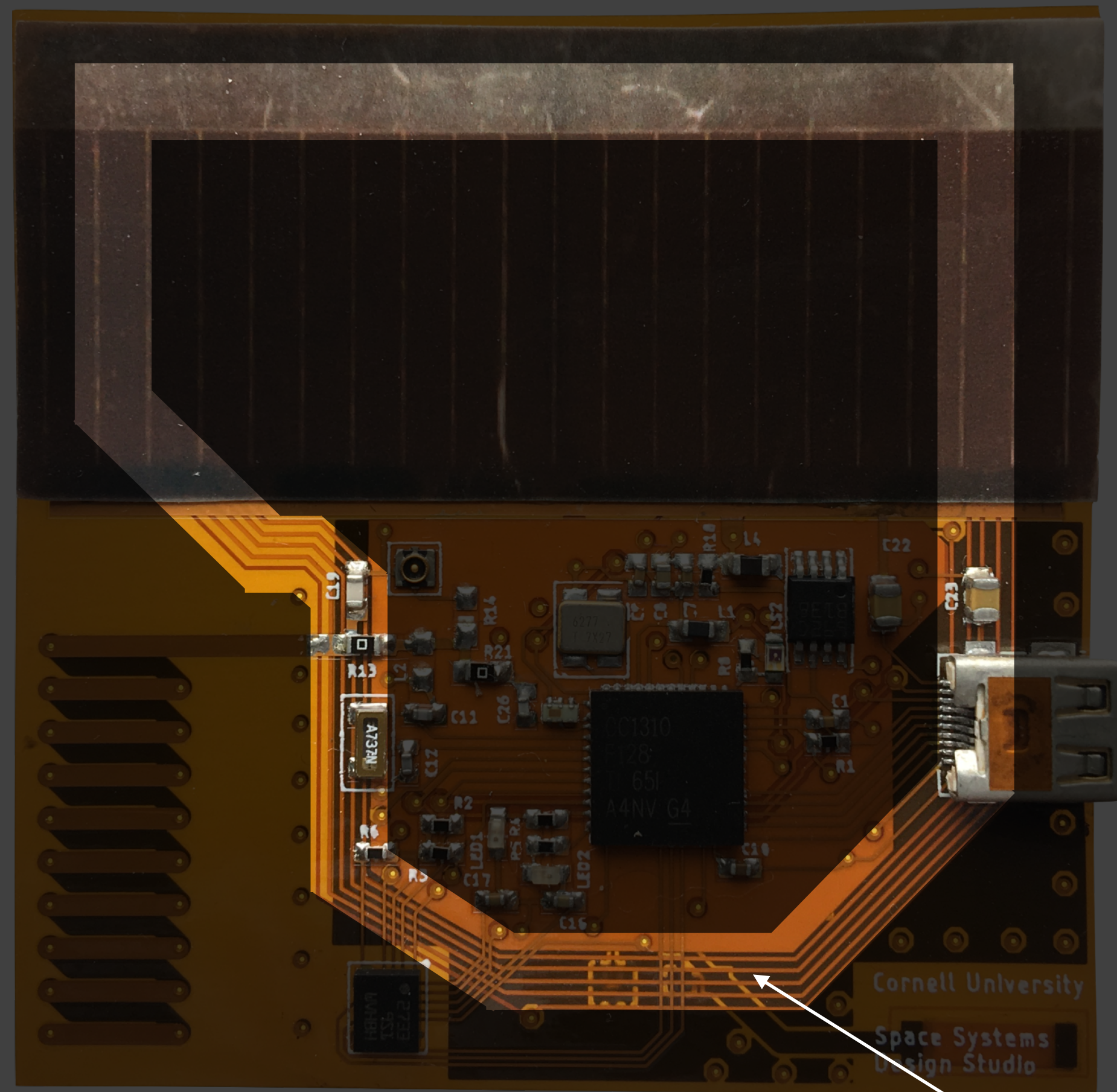
coaxial interfaces to antennas



ARM processor and radio, running real-time operating system

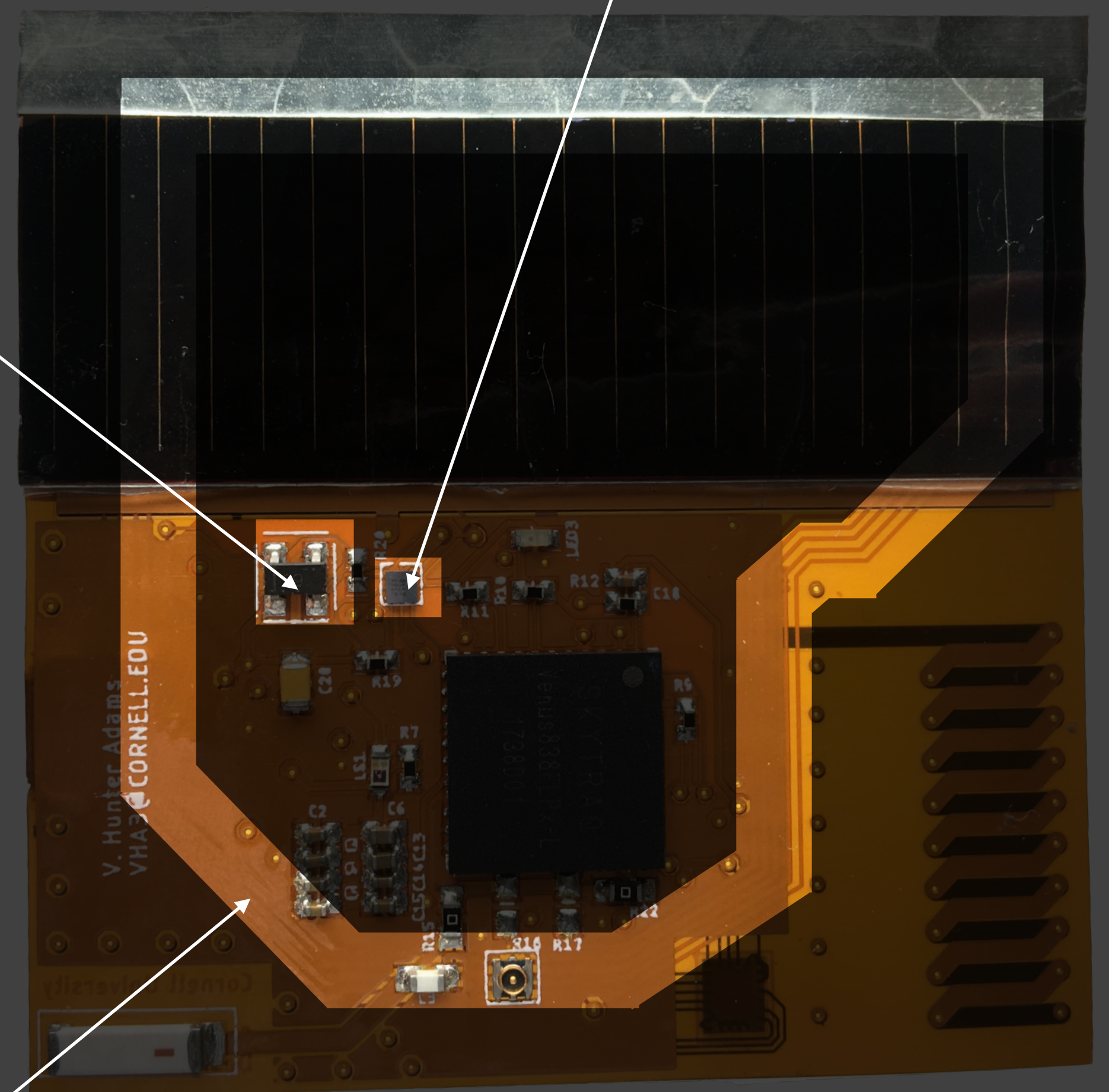


JTAG interface through HDMI port

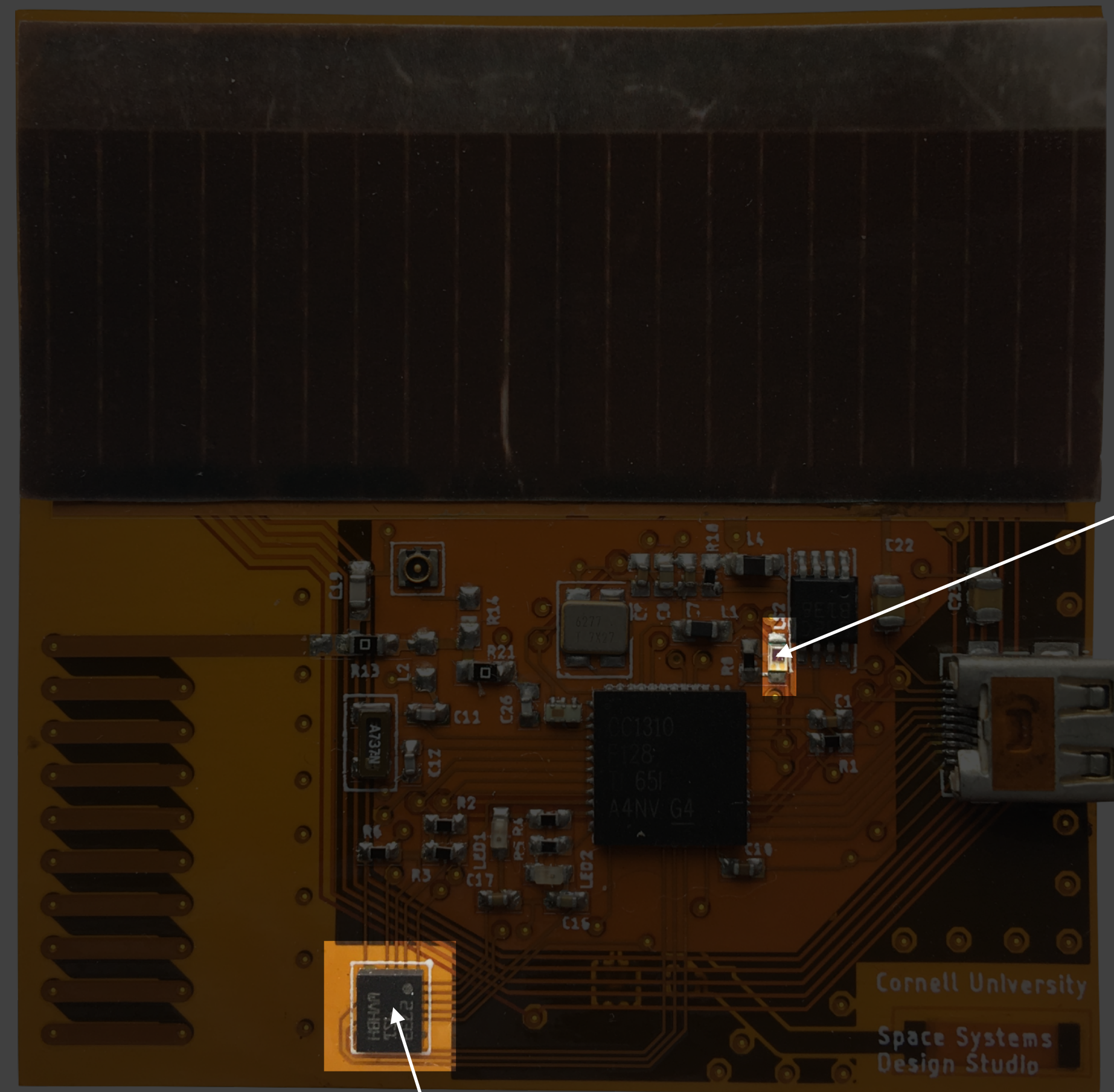


rectifier

motor driver

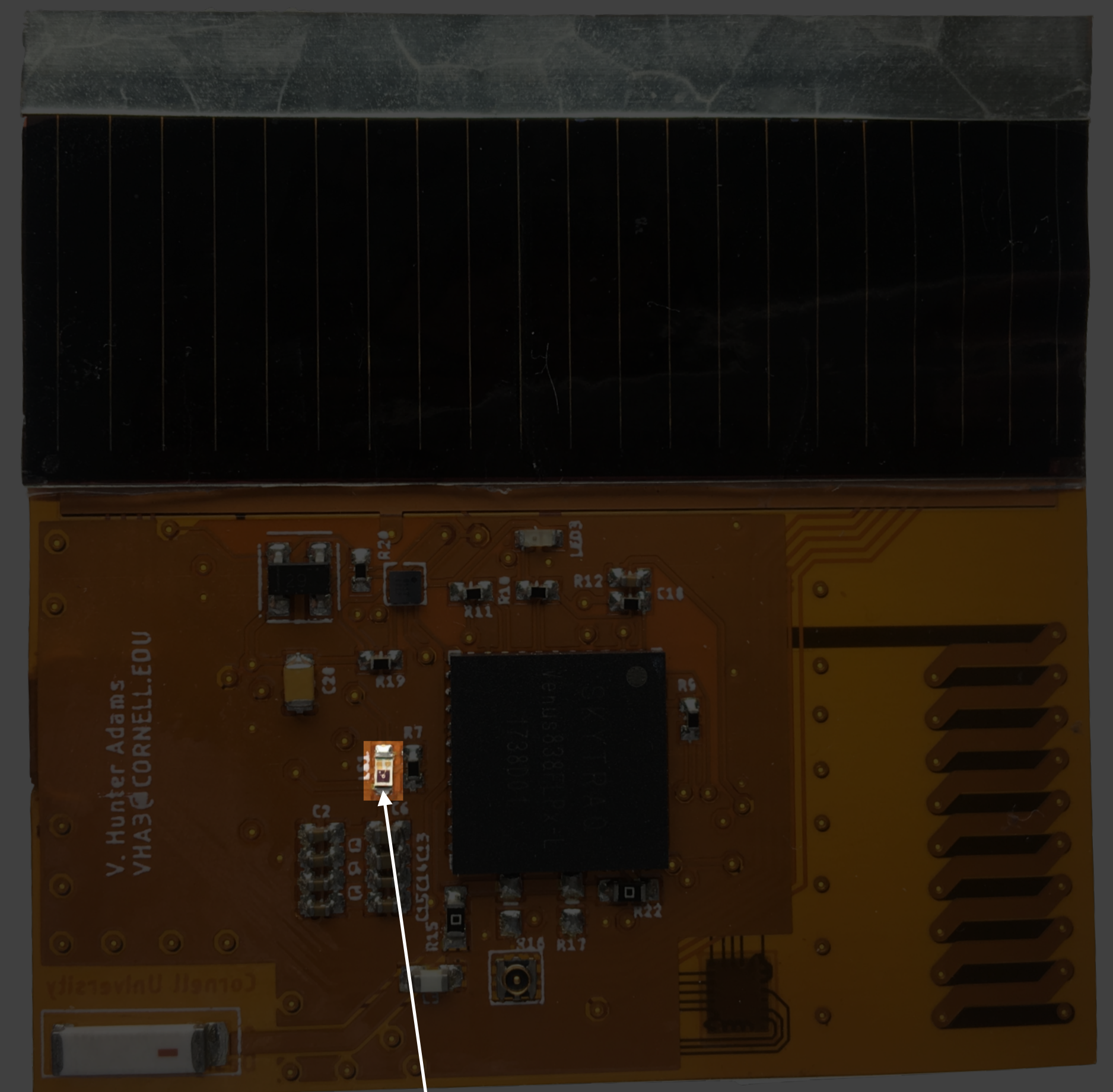


torque/inductive powering coils

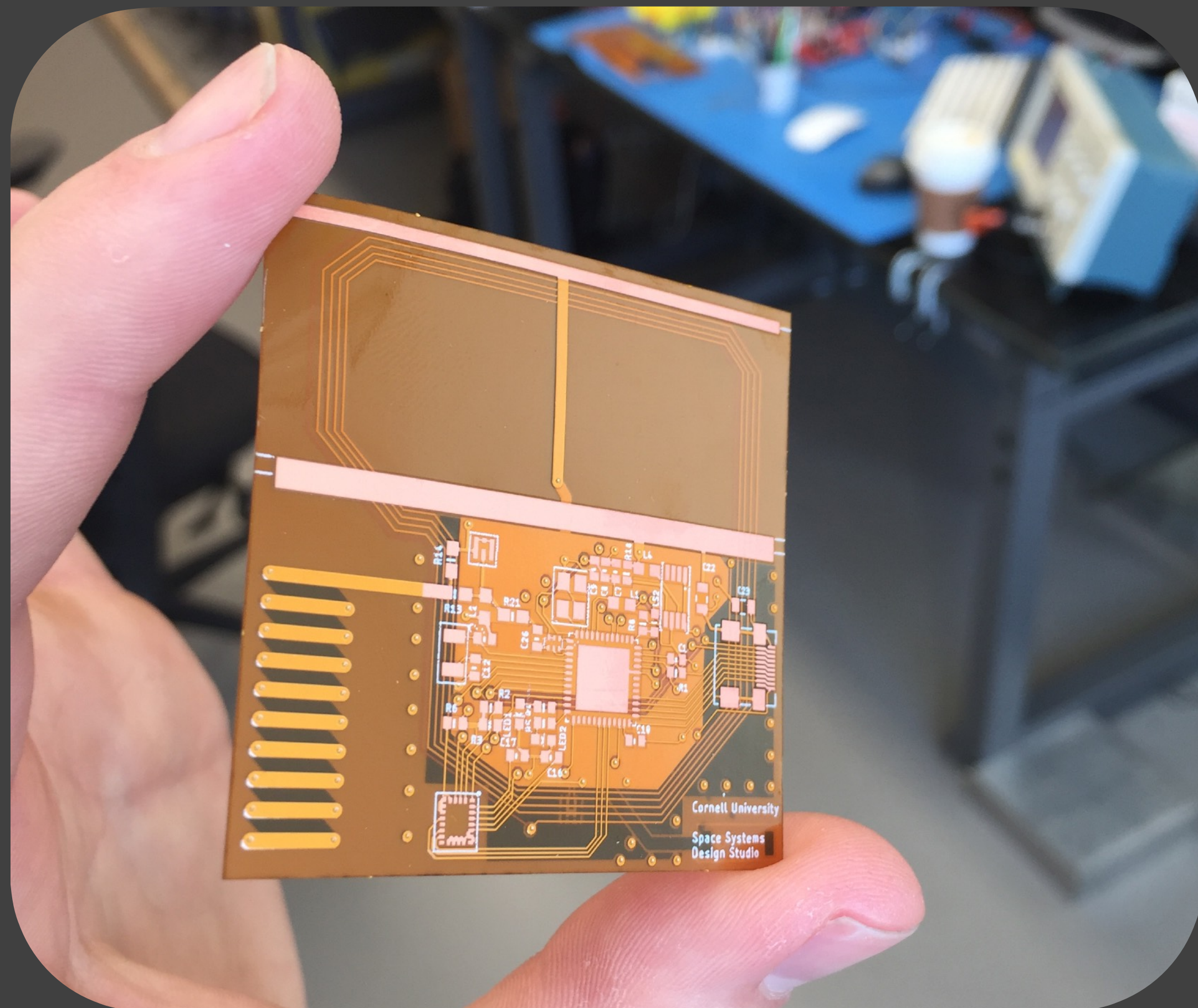


ambient light sensor

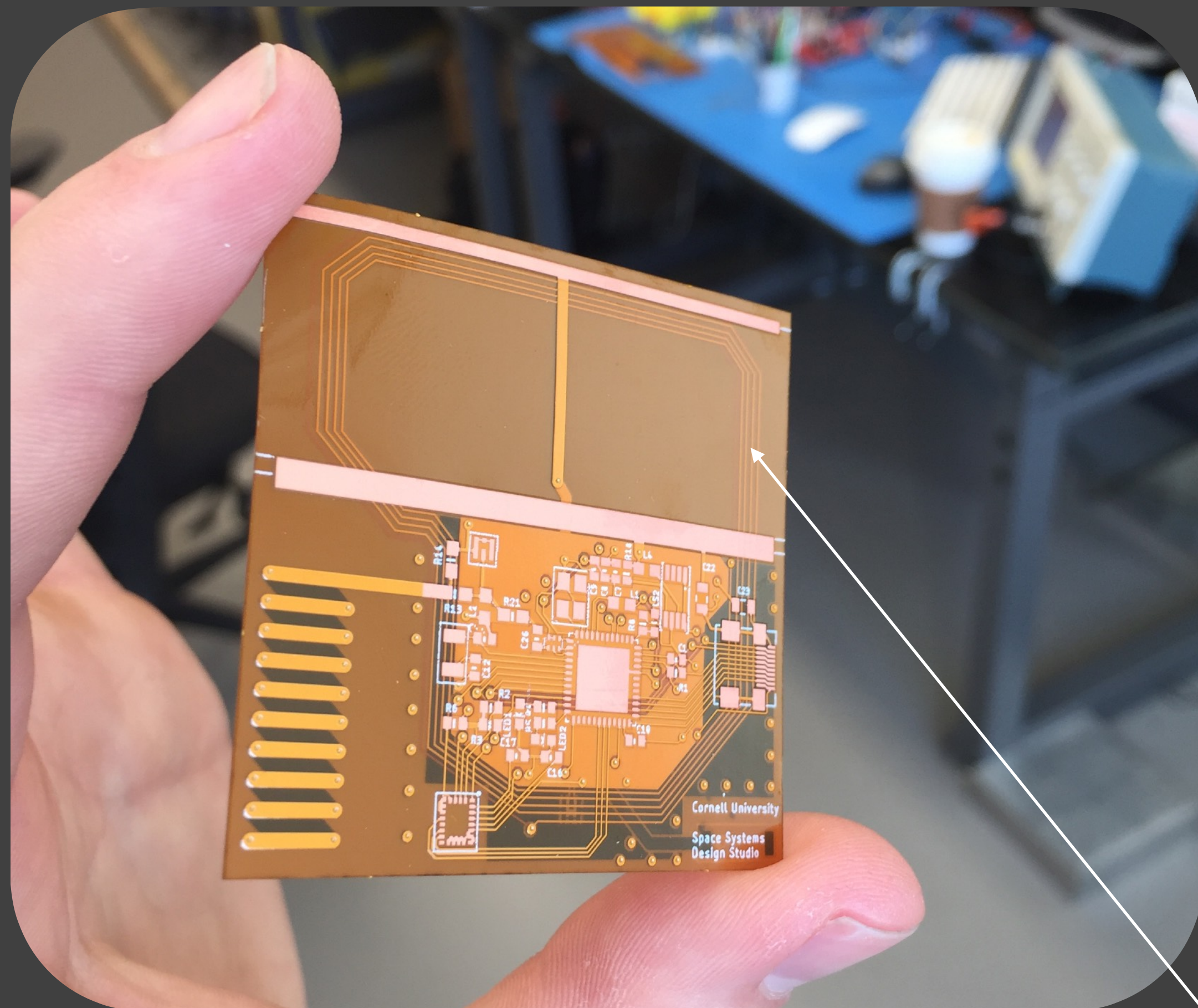
accelerometer, magnetometer, gyroscope, and thermometer



ambient light sensor



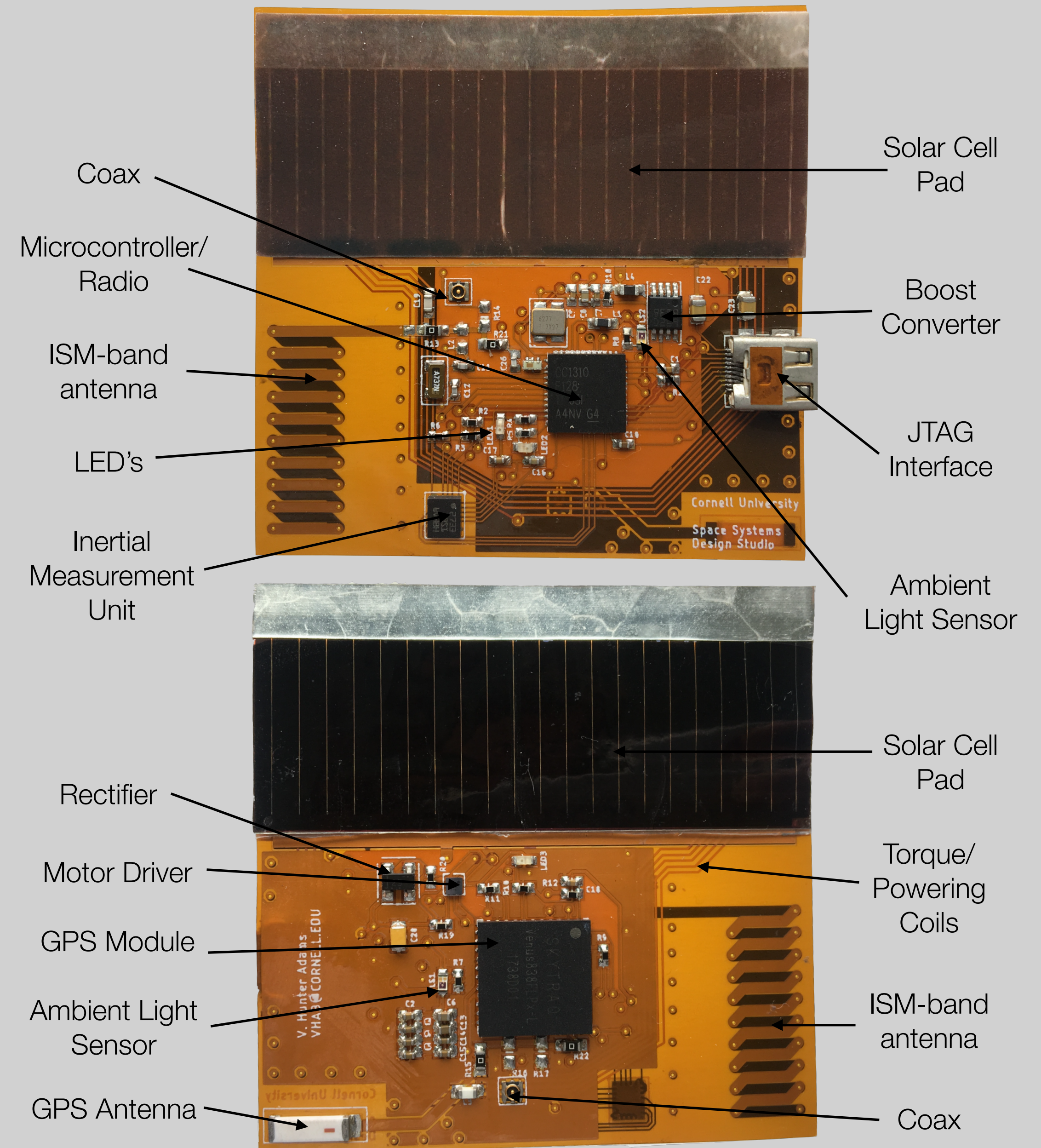
lightweight, flexible Kapton substrate

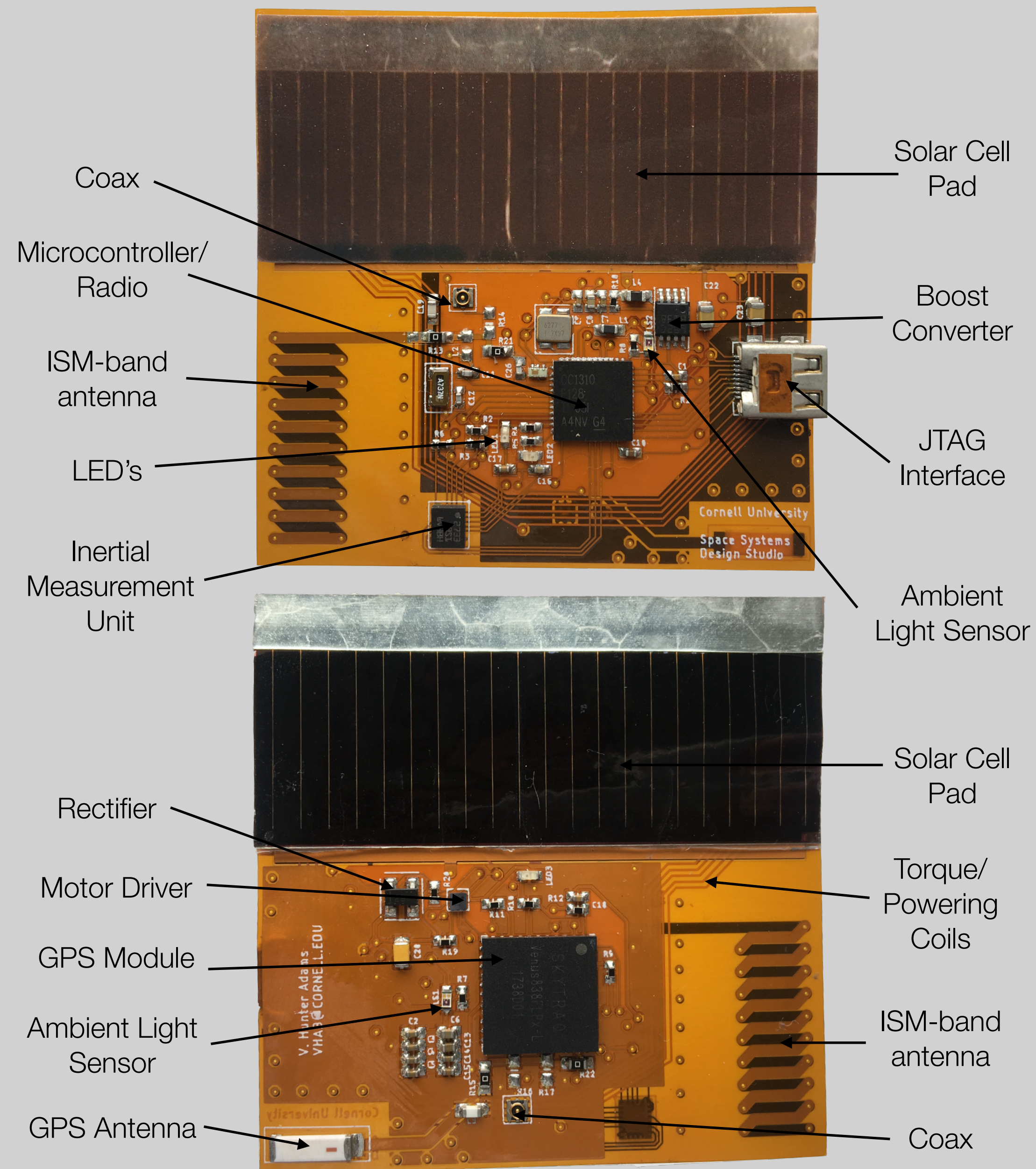


better view of torque/inductive powering coils

Hardware:

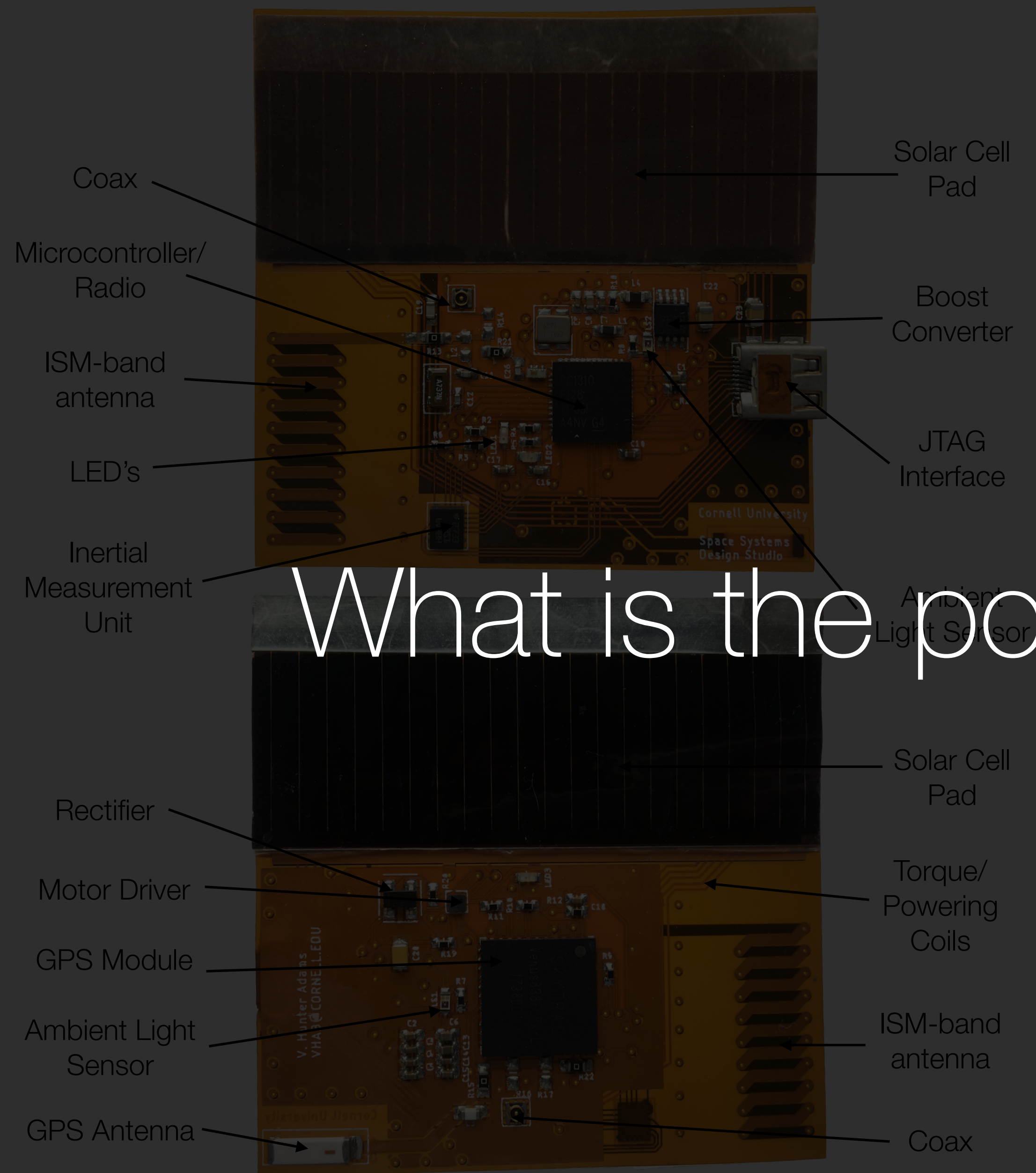
- two Alta-devices solar cells (300 mW each)
- CC1310 ARM processor running RTOS
- 25 mW radio chip
- accelerometer, magnetometer, and gyroscope
- embedded ISM-band antenna (915 MHz)
- GPS
- onboard GPS chip antenna
- JTAG interface via HDMI micro
- two ambient light sensors
- embedded torque coils for attitude manipulation
- motor driver for torque coil control
- embedded inductive powering coils
- LED's for user feedback





Capabilities:

- chip-to-chip communication from up to 1 km line of sight
- chip-to-receiver communication from >1000 km
- GPS acquisition in 30 seconds
- powering by sun and/or inductive coils
- communication among hundreds of chips on a single ISM-band frequency
- can be made to be waterproof
- extremely shock-proof (27,000 g's)
- can generate their own magnetic field
- stable flight in 0 g's
- flexible (to an extent)
- capable of accommodating any sensor that meets size and power requirements
- operating temperatures from -40 to +85 C



What is the point of all this effort?

All Capabilities:

- chip-to-chip communication from up to 1 km line of sight
- chip-to-receiver communication from up to 1000 km
- GPS acquisition in 30 seconds
- powering by sun and/or inductive coils
- communication among hundreds of chips on a single ISM-band frequency
- can be made to be waterproof
- extremely shock-proof (27,000 g's)
- can generate their own magnetic field
- stable flight in 0 g's
- flexible (to an extent)
- capable of accommodating any sensor that meets size and power requirements
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Reasons to build femtosatellites:

- 1.

- 2.

- 3.

Reasons to build femtosatellites:

1. They can be manufactured cheaply and in bulk.
- 2.
- 3.

Reasons to build femtosatellites:

1. They can be manufactured cheaply and in bulk.
2. They can be launched and deployed in bulk.
- 3.

Reasons to build femtosatellites:

1. They can be manufactured cheaply and in bulk.
2. They can be launched and deployed in bulk.
3. Their size affects their orbital perturbations and impact mechanics.

Reasons to build femtosatellites:

1. They can be manufactured cheaply and in bulk.
2. They can be launched and deployed in bulk.
3. Their size affects their orbital perturbations and impact mechanics.

Femtosatellites are *not* small versions of large spacecraft. They have a different set of use cases.

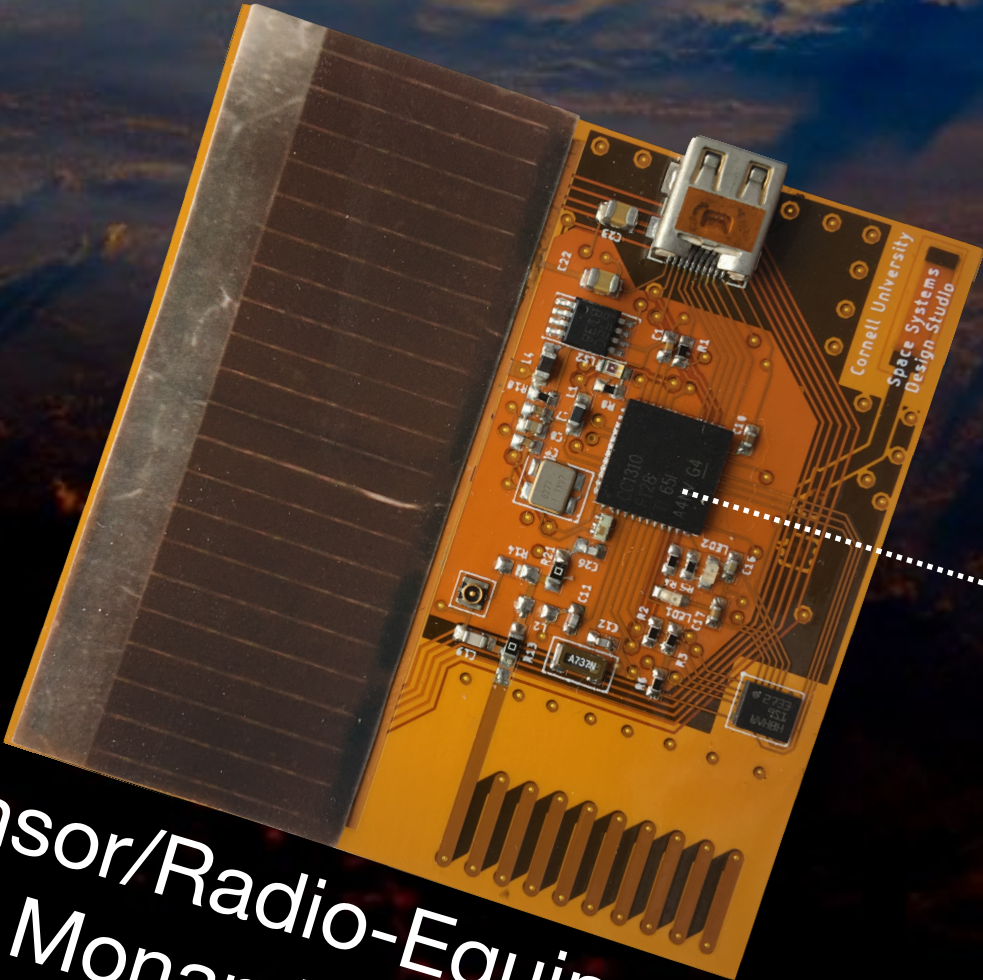
Duration of the presentation:

1. Short-term mission possibilities
2. Intermediate-term mission possibilities
3. Long-term speculation

Ground Stations,
Aggregating Data



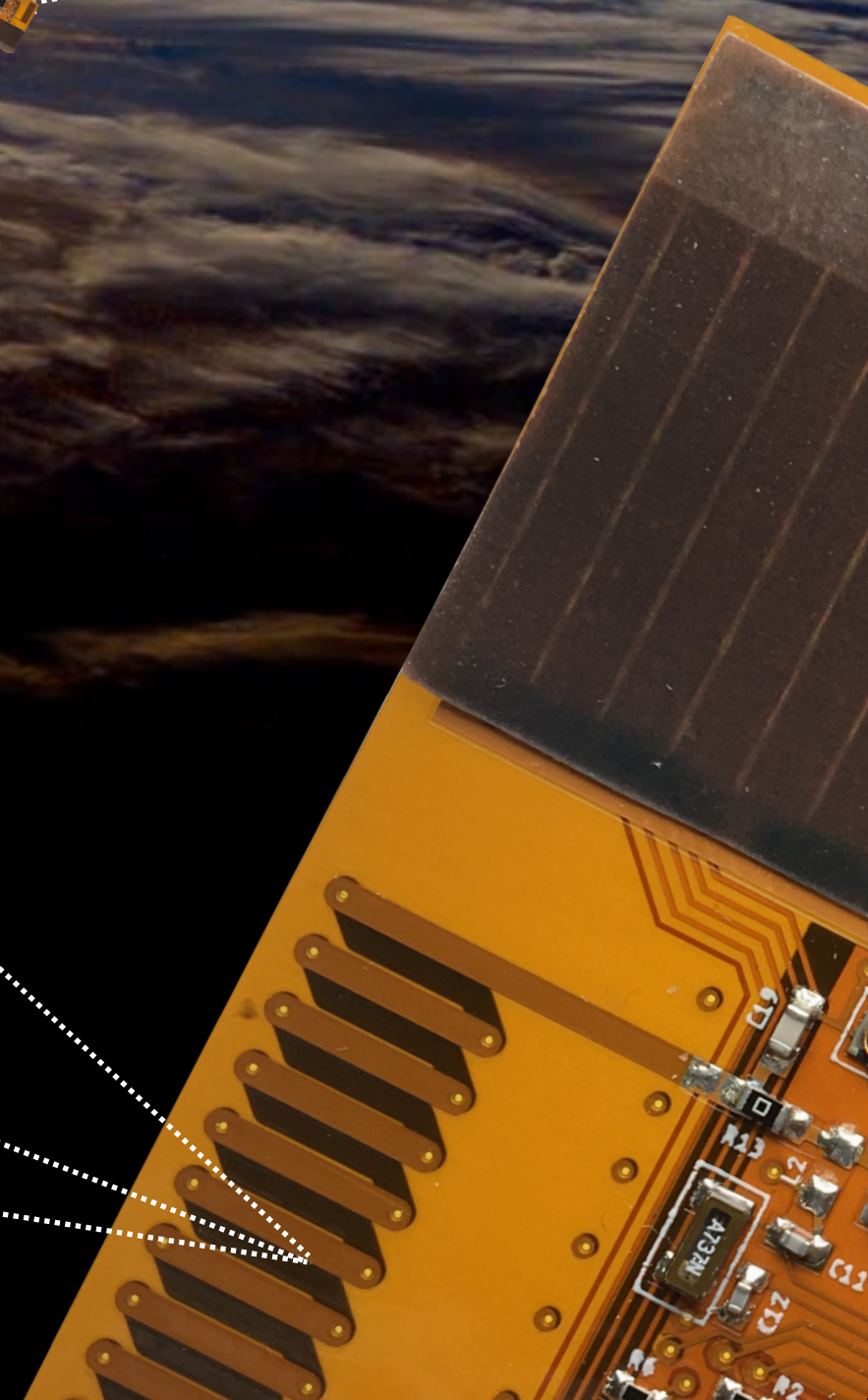
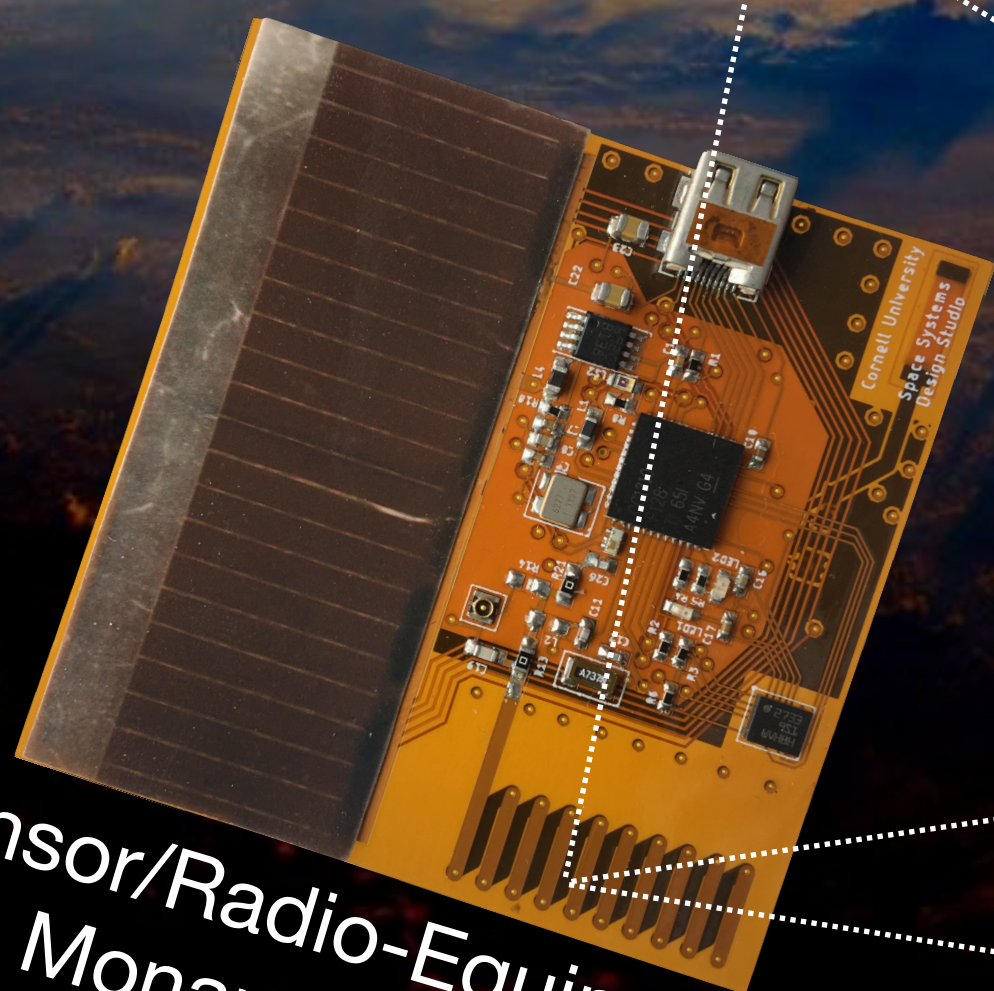
Sensor/Radio-Equipped
Monarch Node



Ground
Stations, Aggregating
Data and Distributing
Swarm Commands



Sensor/Radio-Equipped
Monarch Node



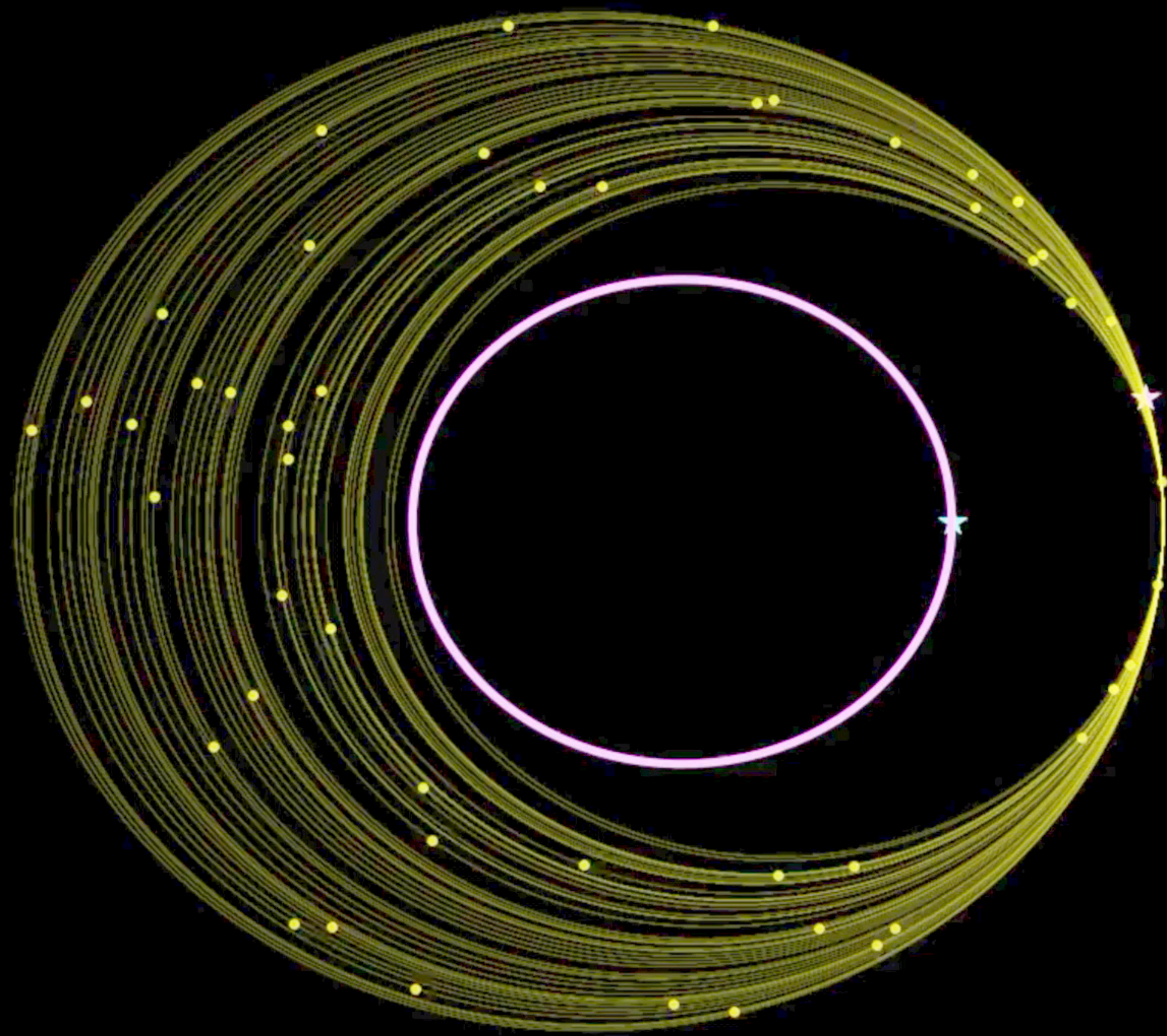
Ground
Stations, Aggregating
Data and Distributing
Swarm Commands

Technical requirements:

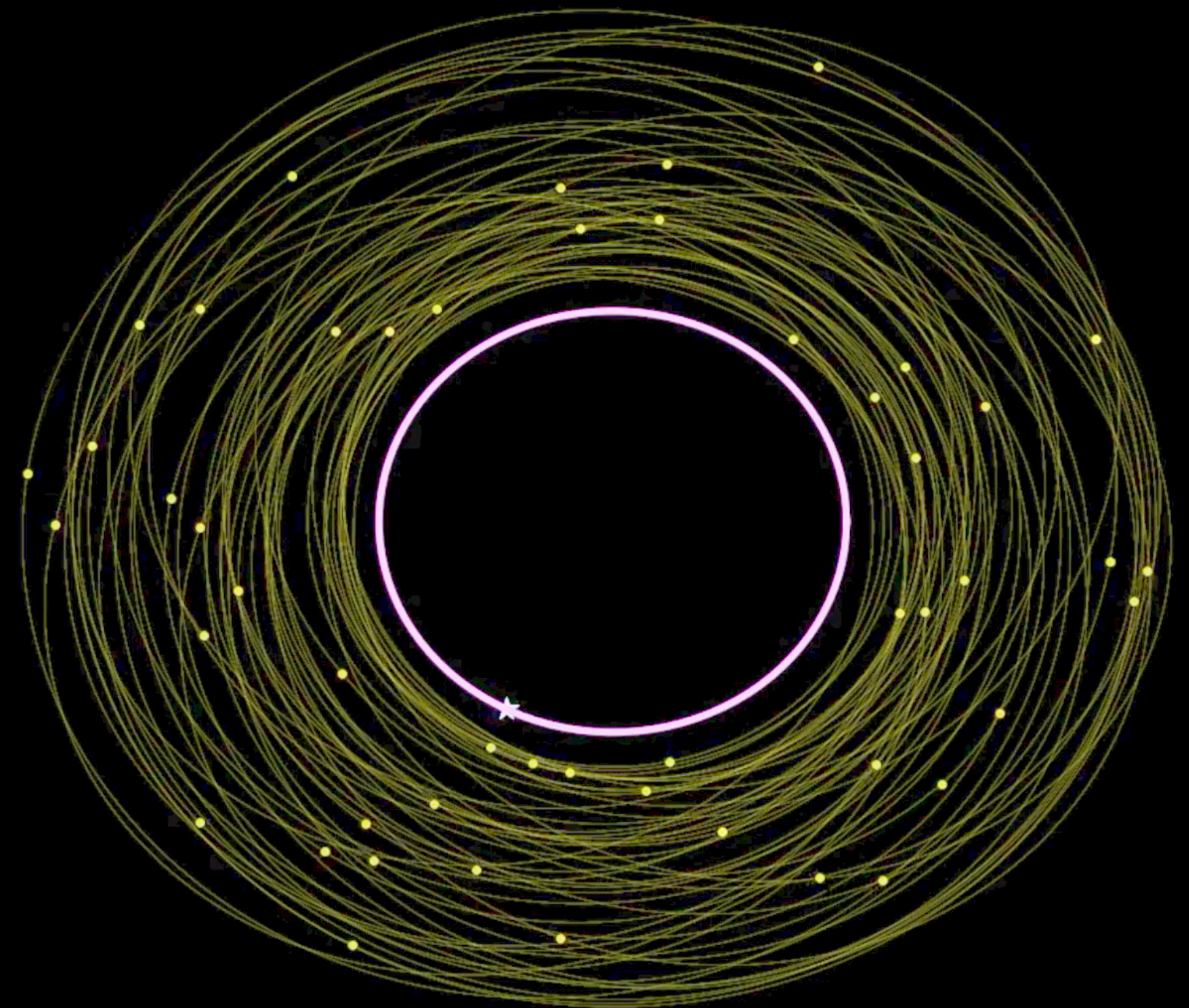
1. Routing policy over the collection of Monarchs
2. Information exchange among the Monarchs

Sensor/Radio-Equipped
Monarch Node

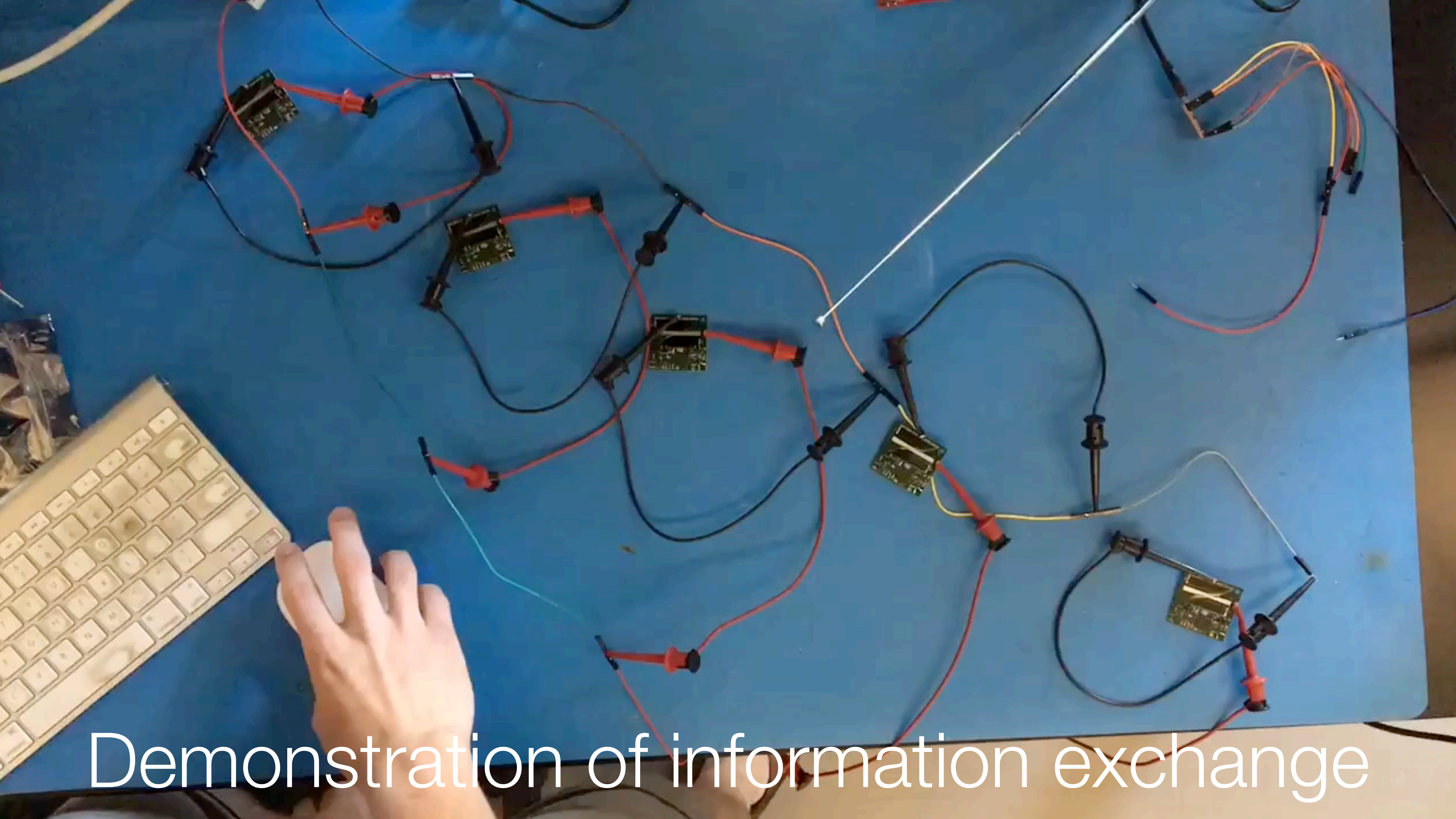
Simulation of proposed routing policy



routing over nested swarm



routing over stochastic swarm

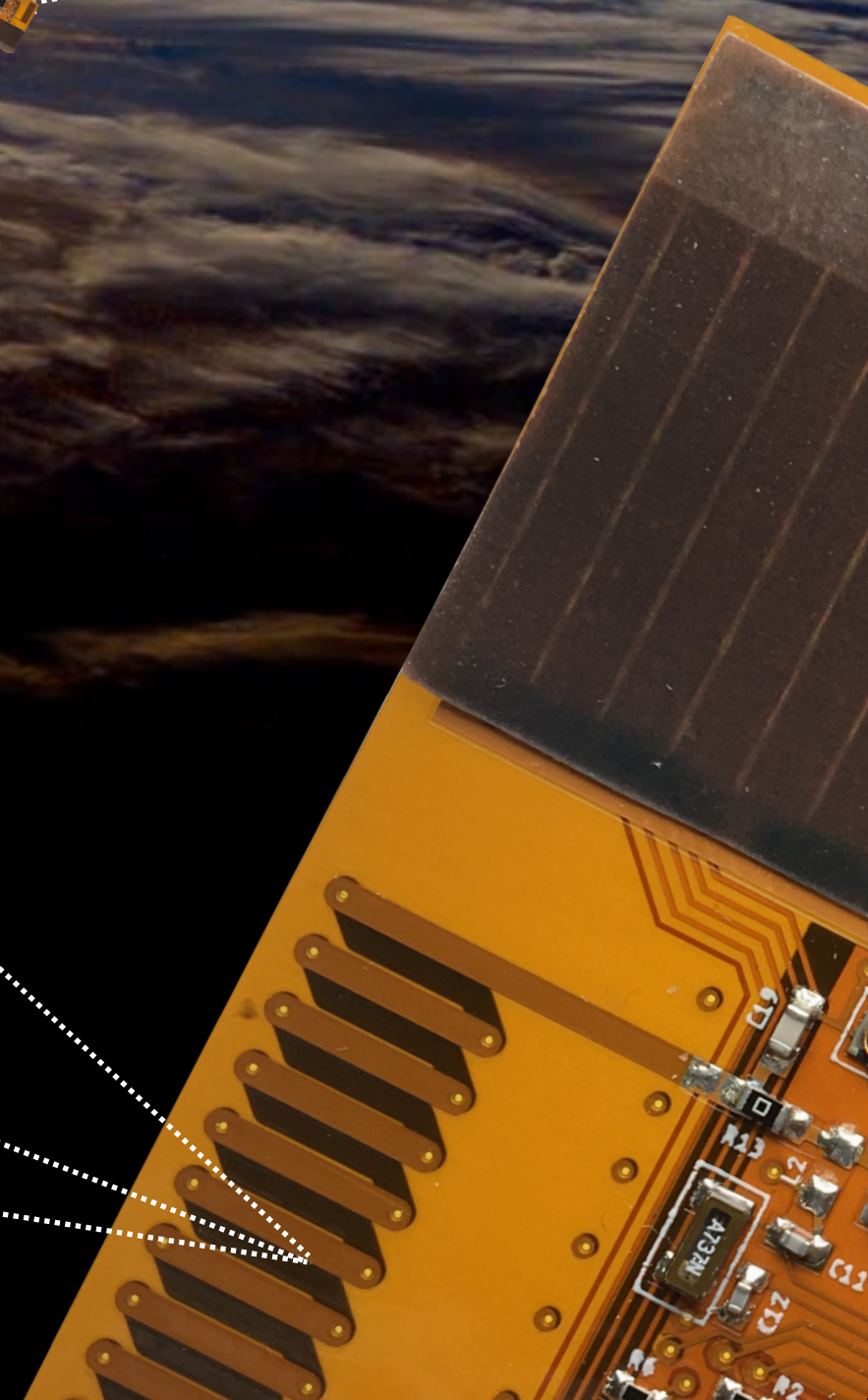
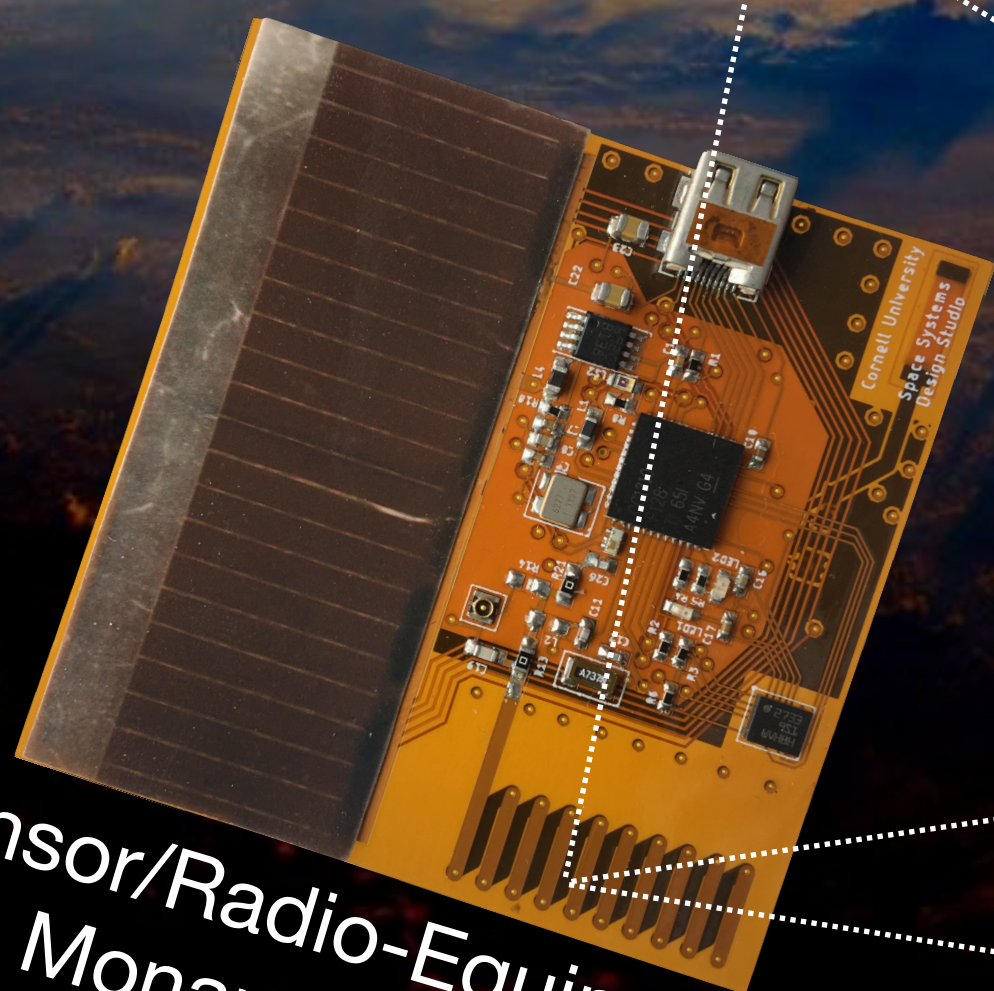


Demonstration of information exchange

Ground
Stations, Aggregating
Data and Distributing
Swarm Commands

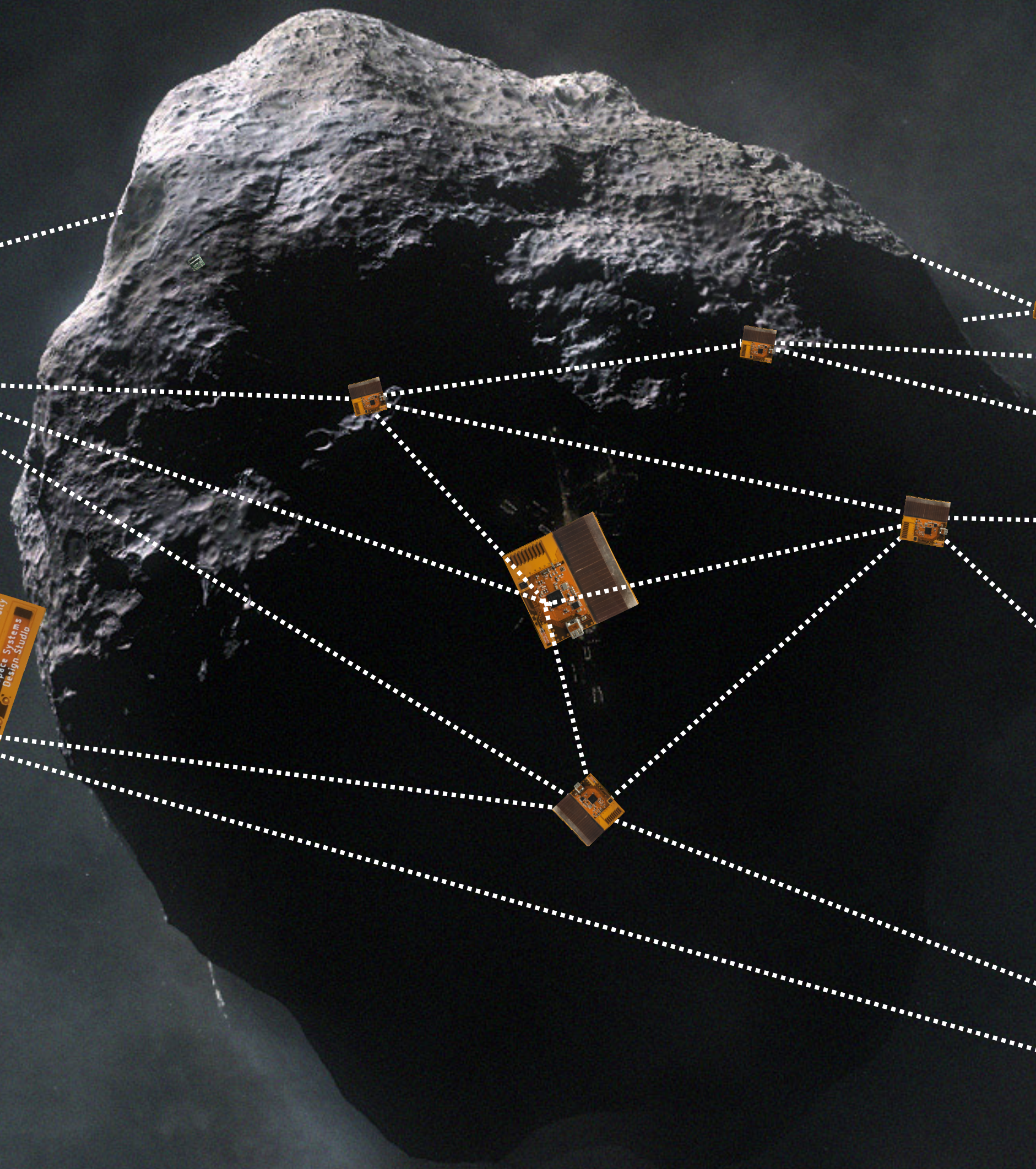


Sensor/Radio-Equipped
Monarch Node

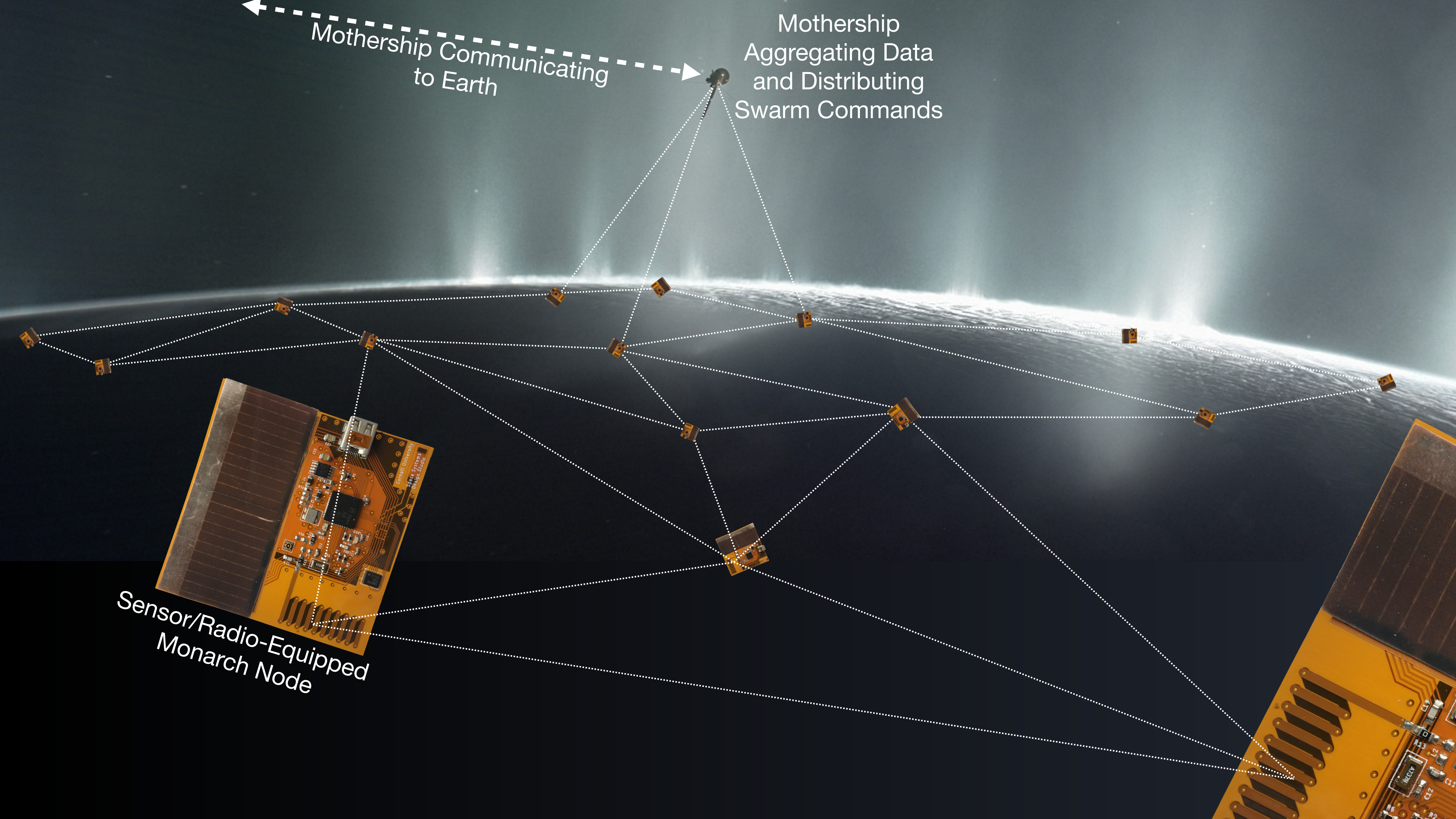




Sensor/Radio-Equipped
Monarch Node



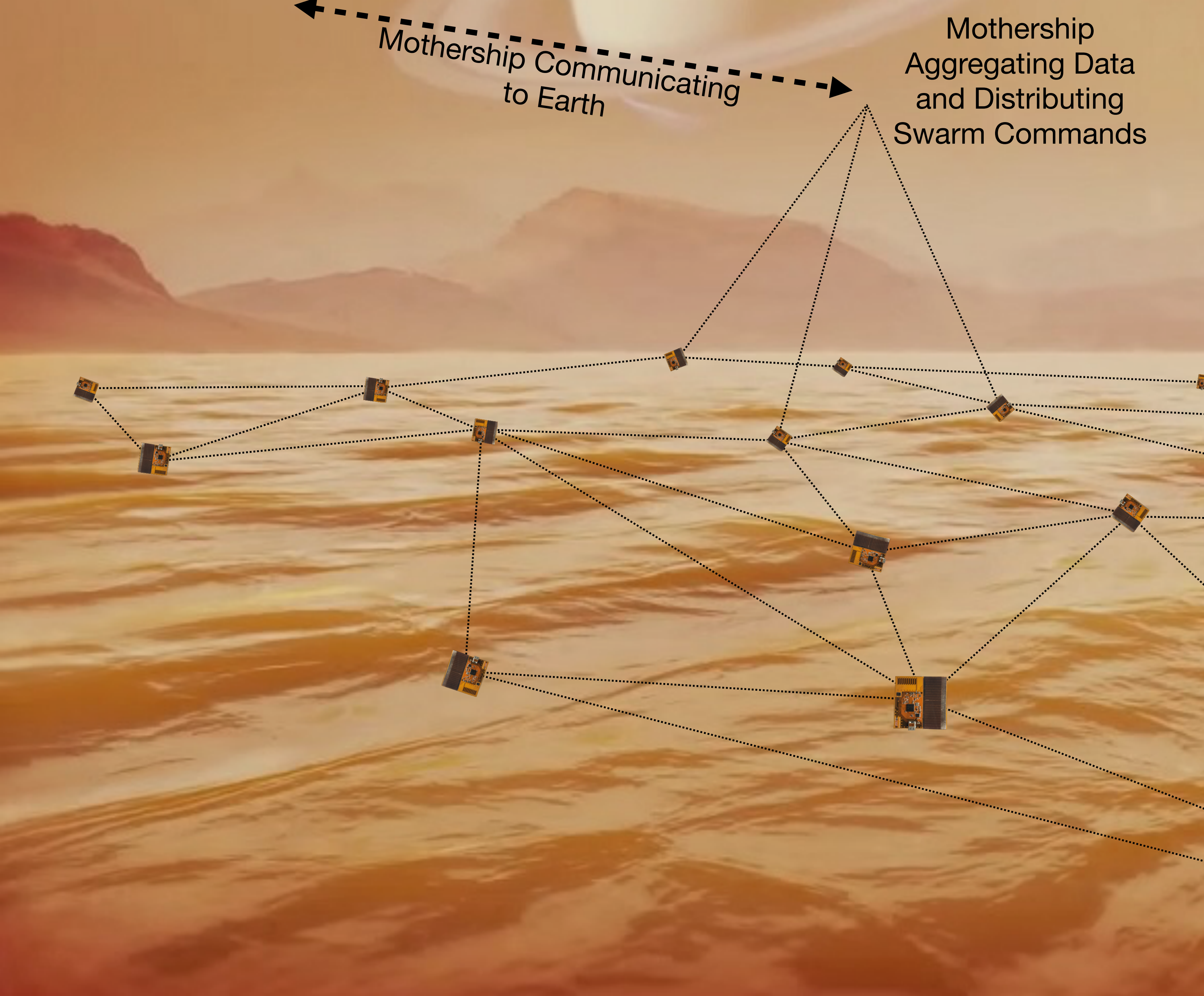
Sensor/Radio-Equipped
Monarch Node

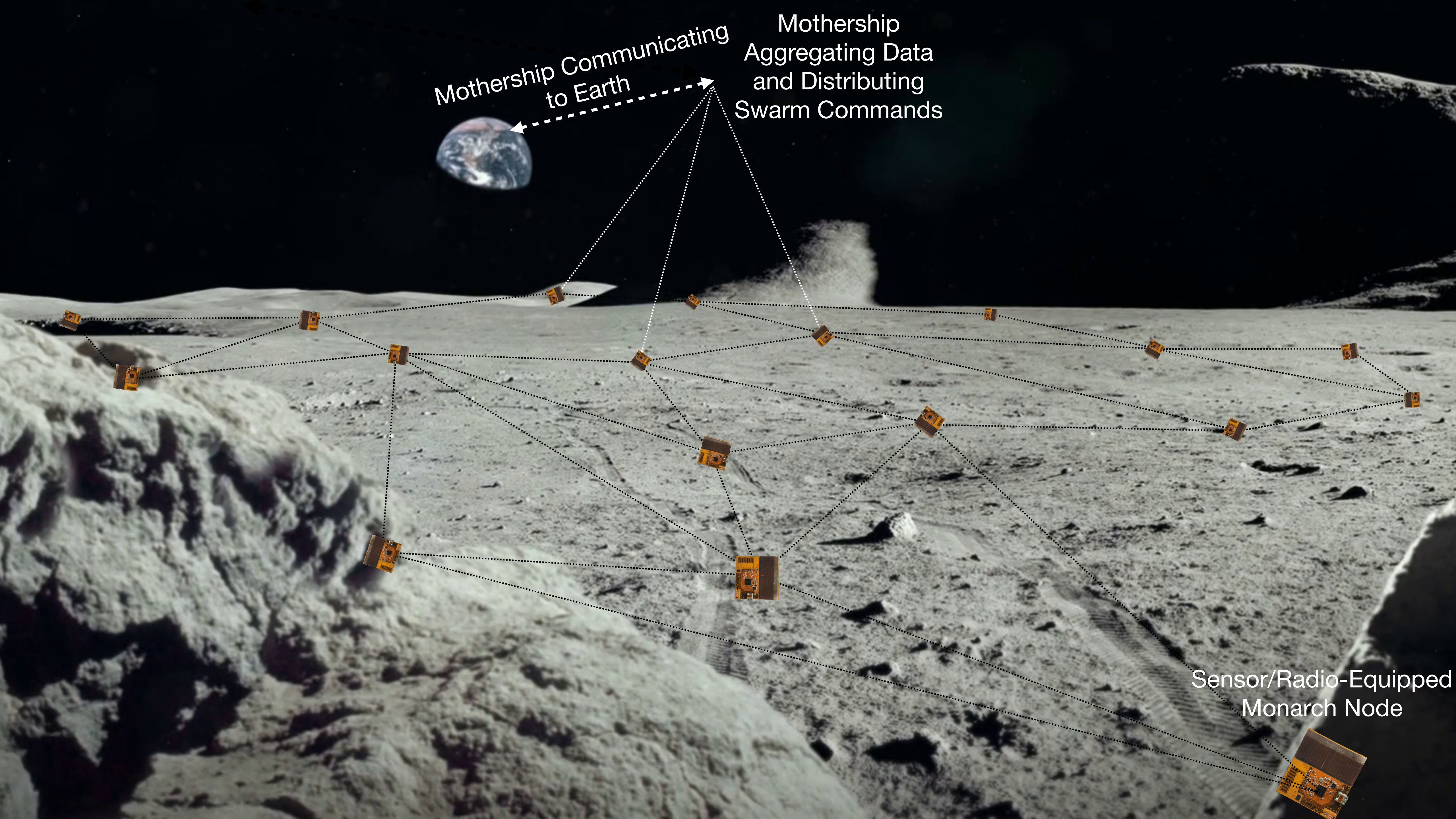


Mothership Communicating to Earth

Mothership
Aggregating Data
and Distributing
Swarm Commands

Sensor/Radio-Equipped
Monarch Node





Mothership Communicating
to Earth

Mothership
Aggregating Data
and Distributing
Swarm Commands

Sensor/Radio-Equipped
Monarch Node

Mothership Communicating
to Earth

Mothership
Aggregating Data
and Distributing
Swarm Commands

Would the spacecraft survive impact?

Sensor/Radio-Equipped
Monarch Node

Lunar Impact Survivability Testing

5000 - 27,000 g's normal
to the board surface



lunar regolith simulant

Lunar Impact Survivability Testing

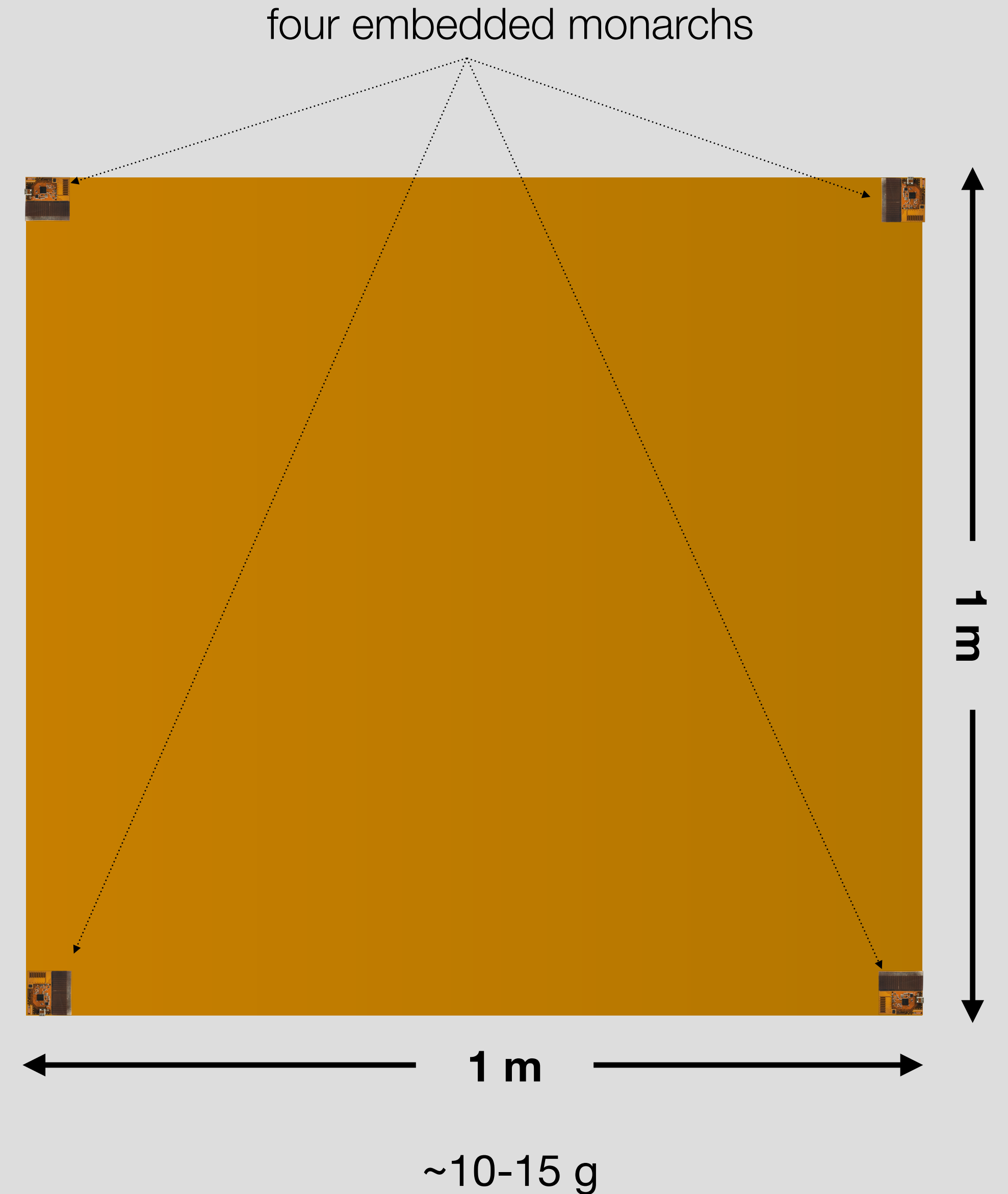
5000 - 27,000 g's normal
to the board surface

All boards and sensors survived unscathed.

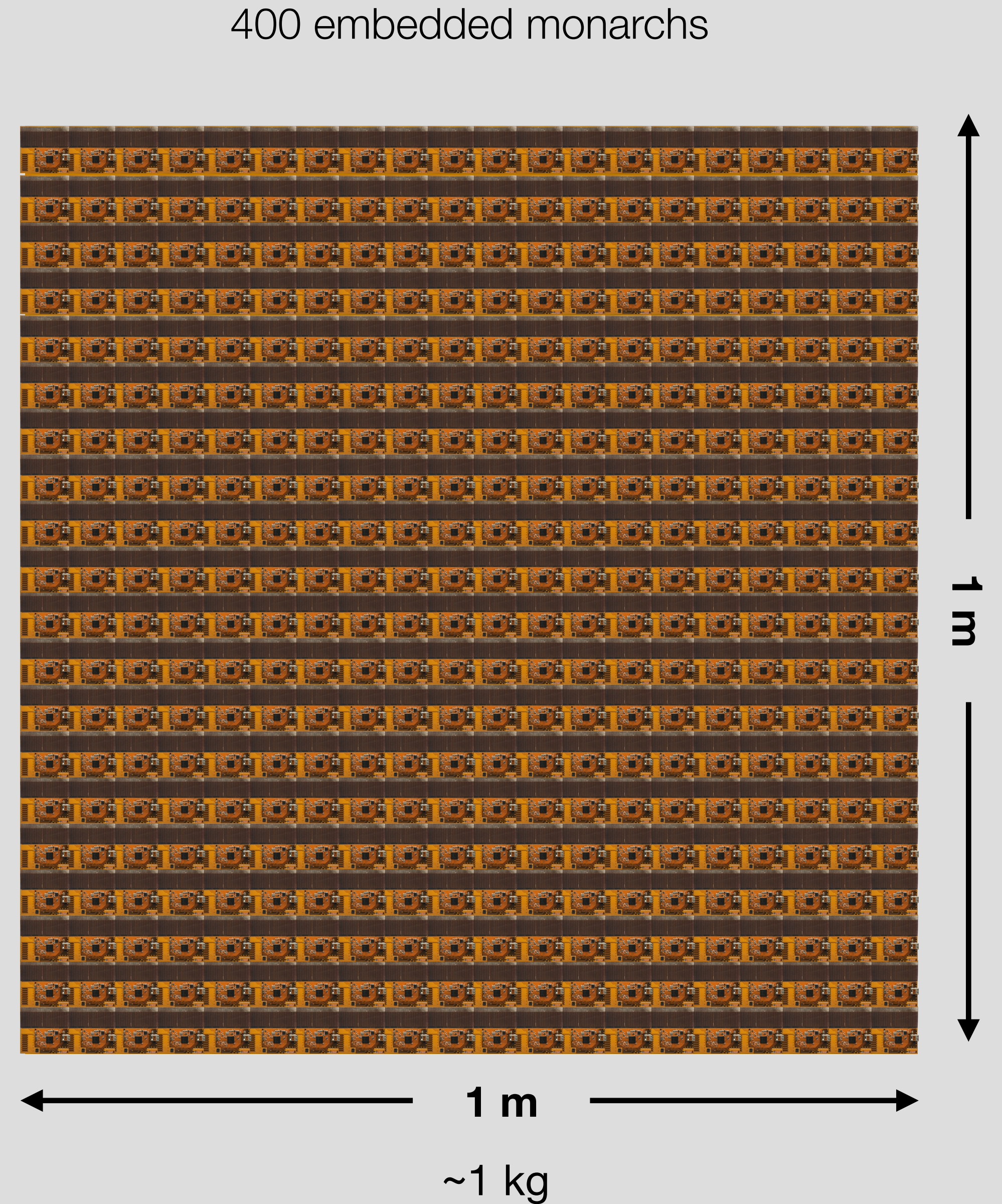
lunar regolith simulant



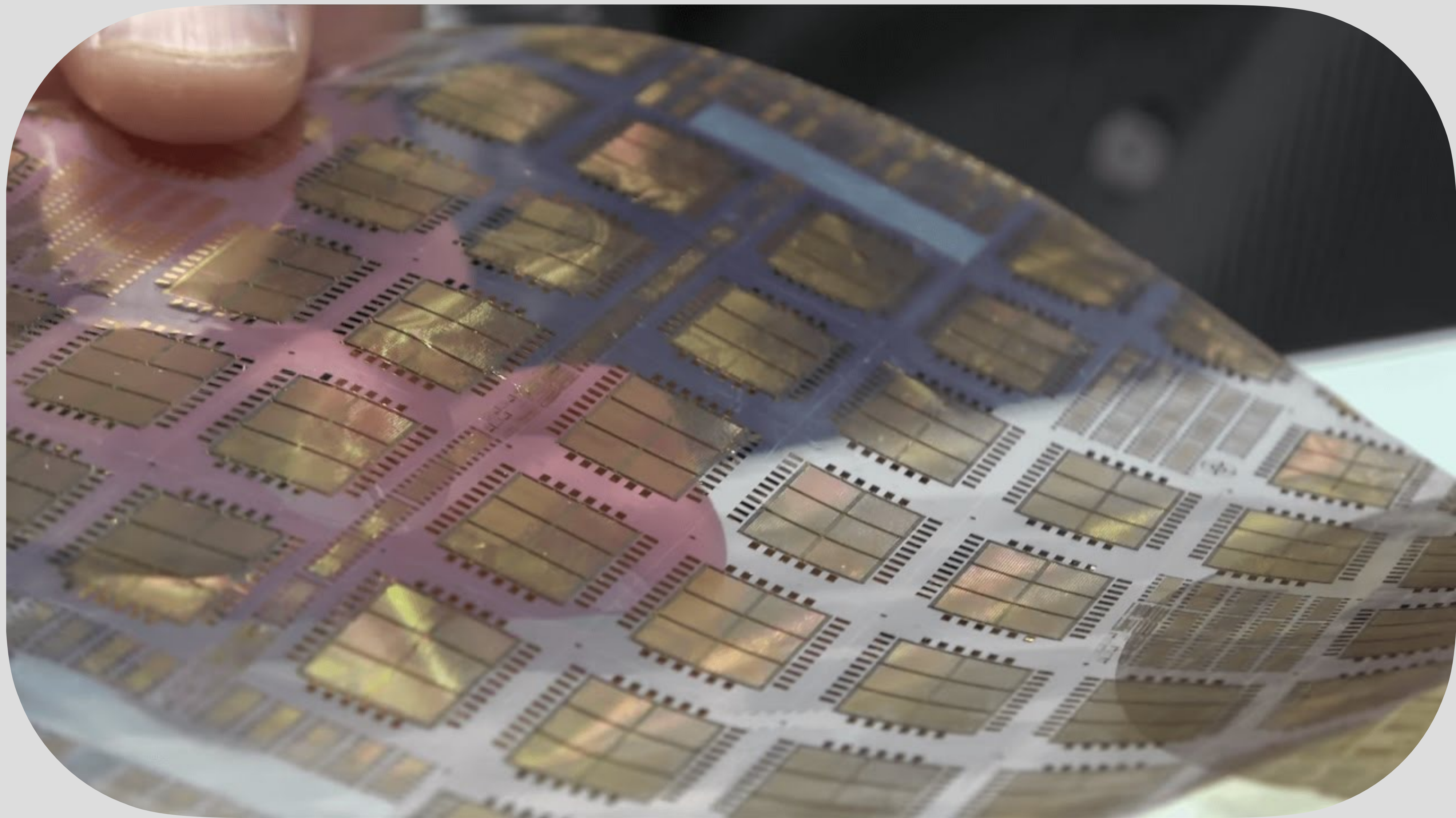
Rather than attach a sail to a spacecraft, we can create a sail that is a spacecraft.



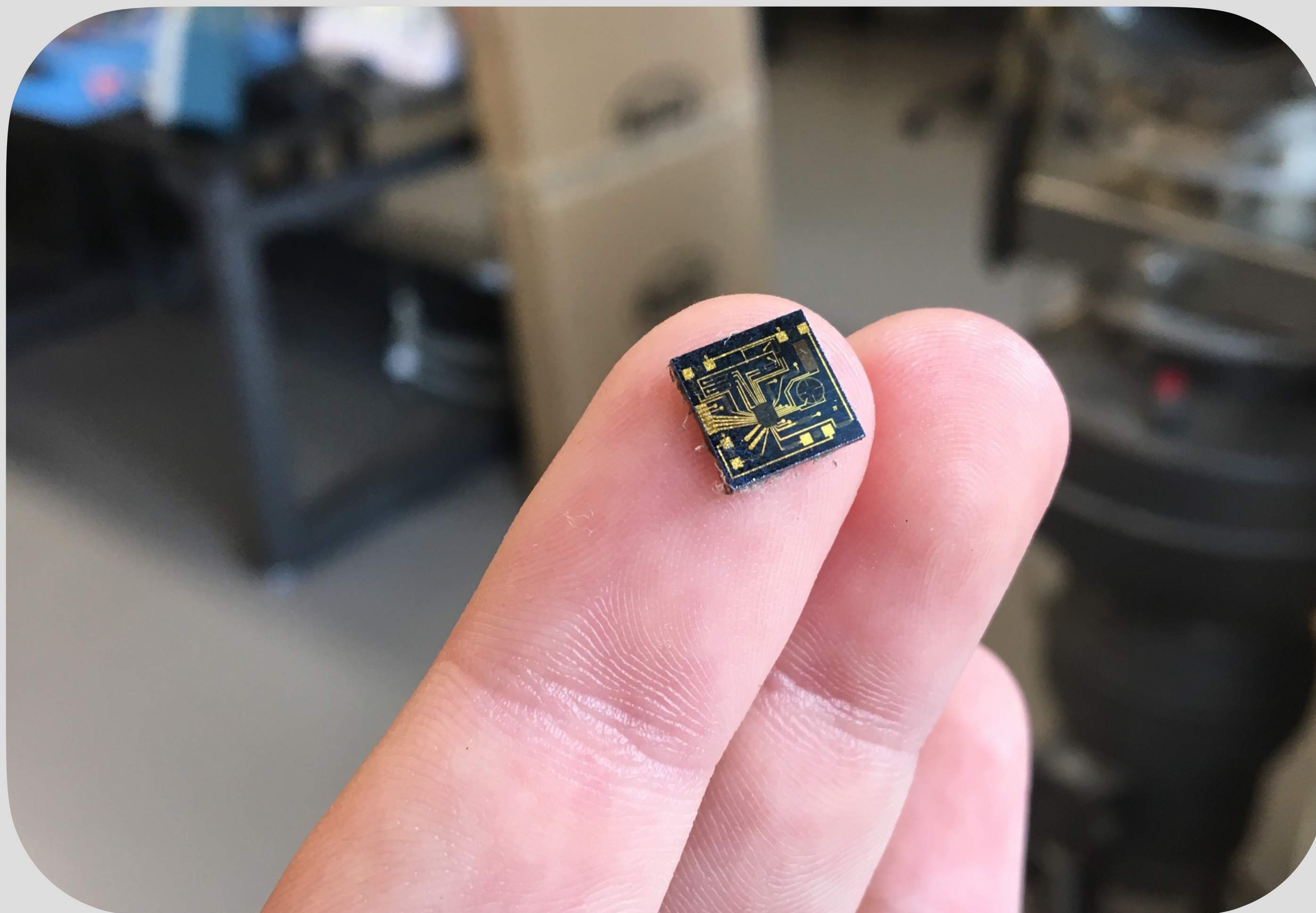
Or a sail that is
composed of 400
spacecraft, with 400
sensor suites.



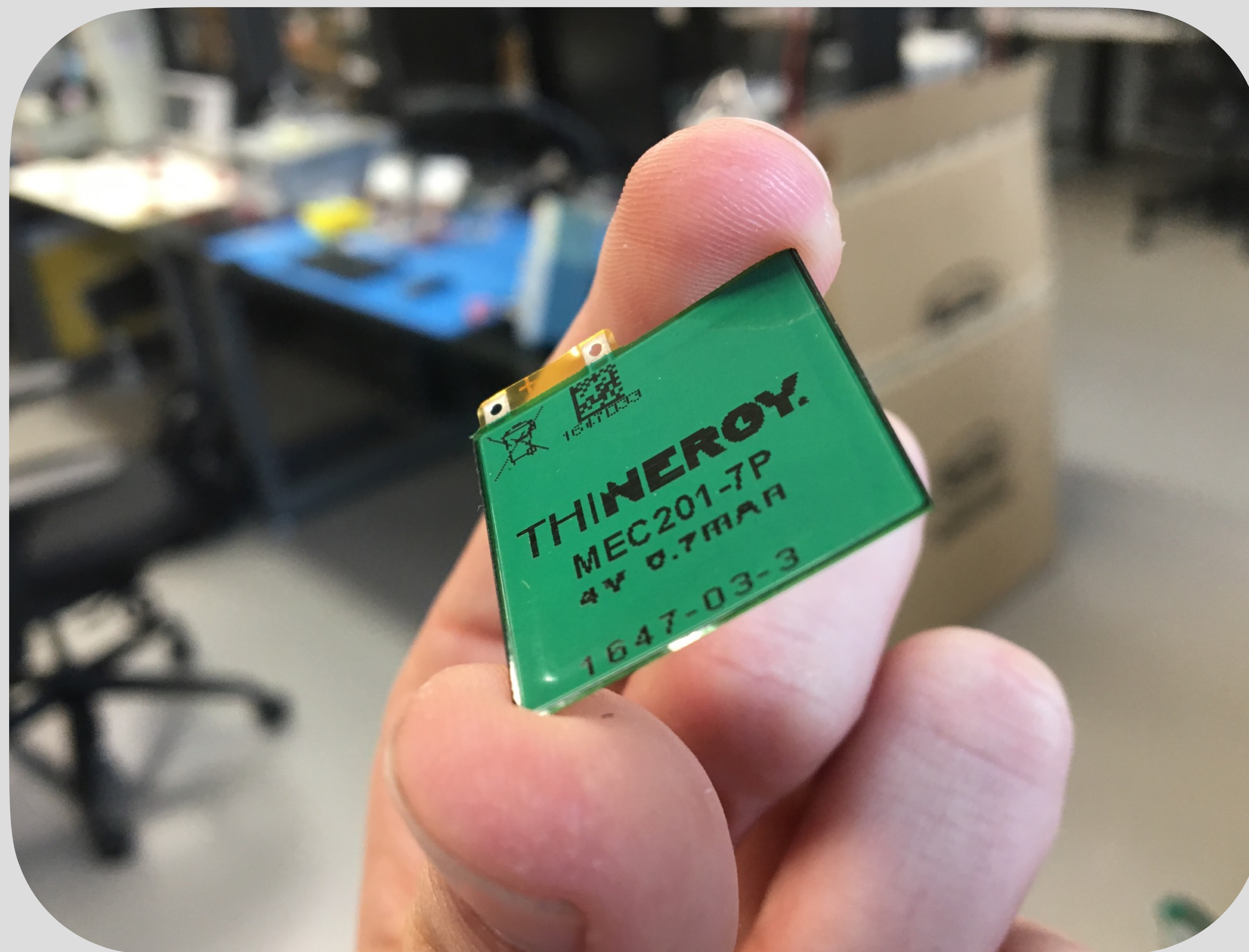
Long-term speculation . . .



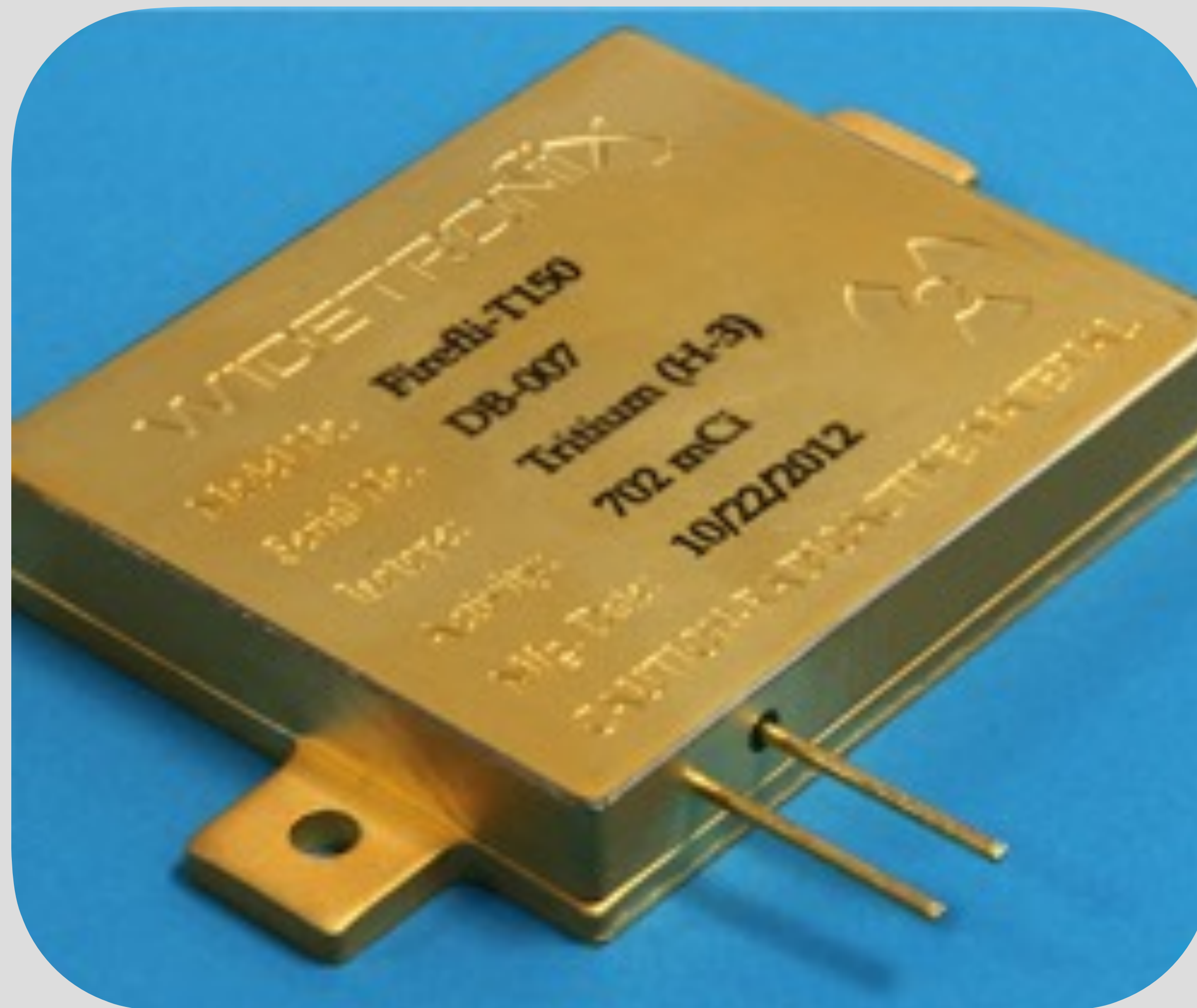
PragmatlC thin-film ARM processors



Starchip mockup



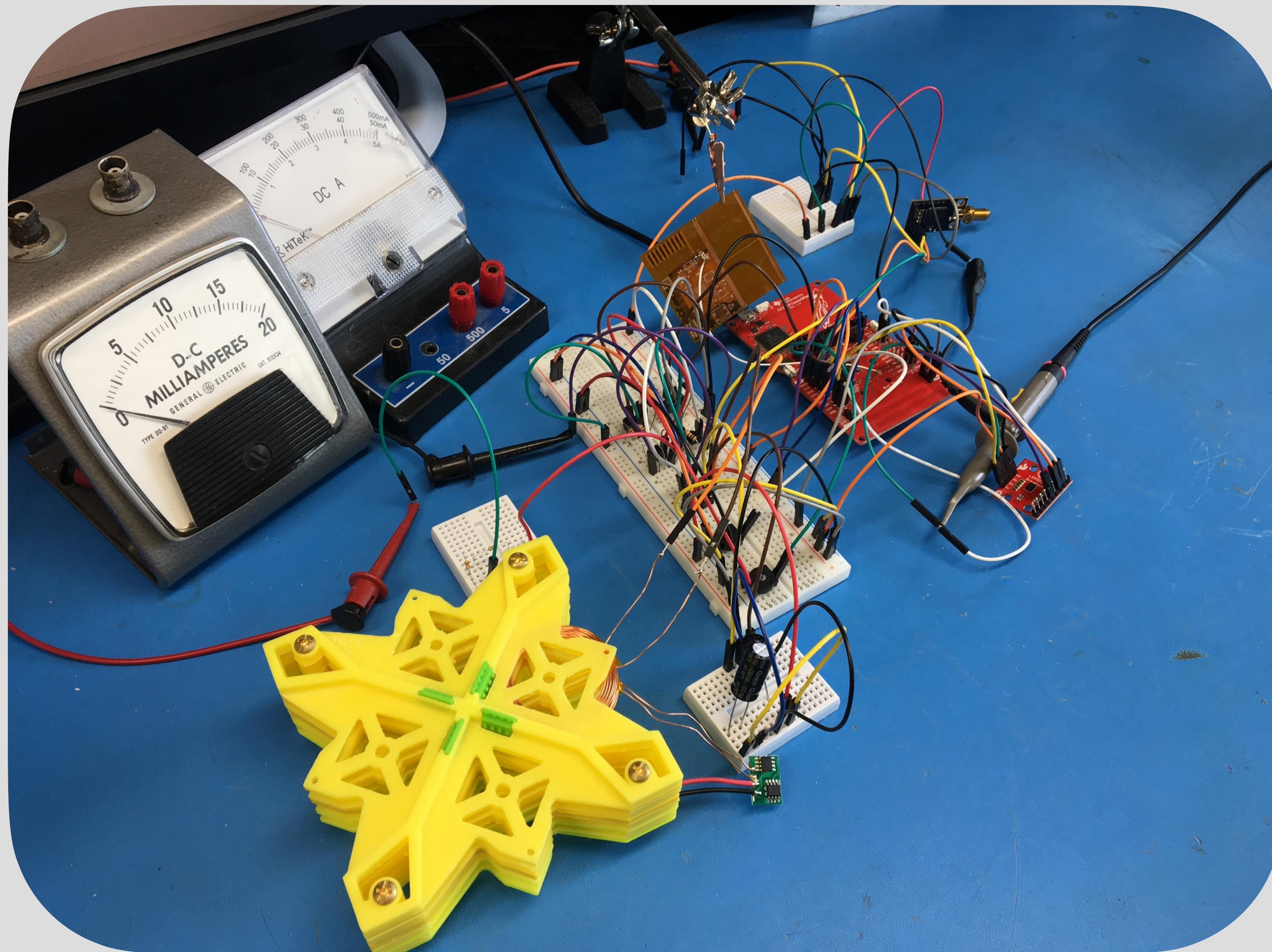
Thineroy solid-state battery



Widetrnix beta voltaic nuclear battery



Emergent pattern in a murmuration of starlings



My lab bench